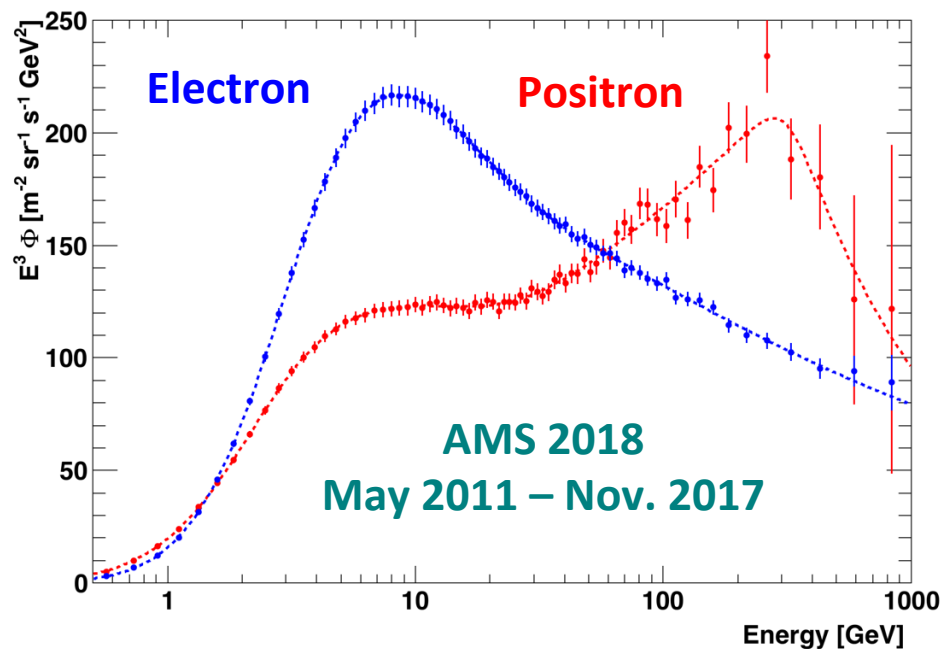
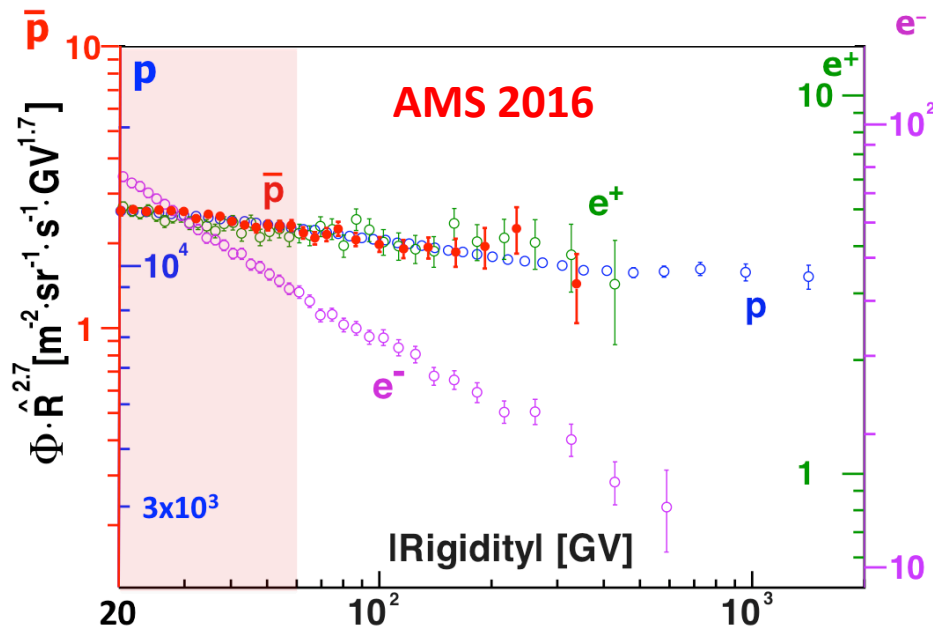
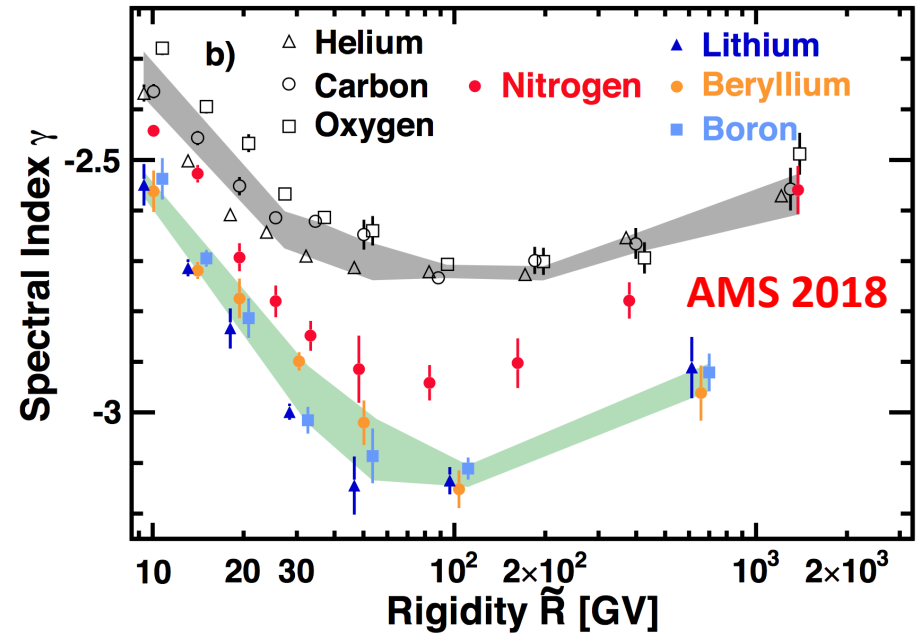
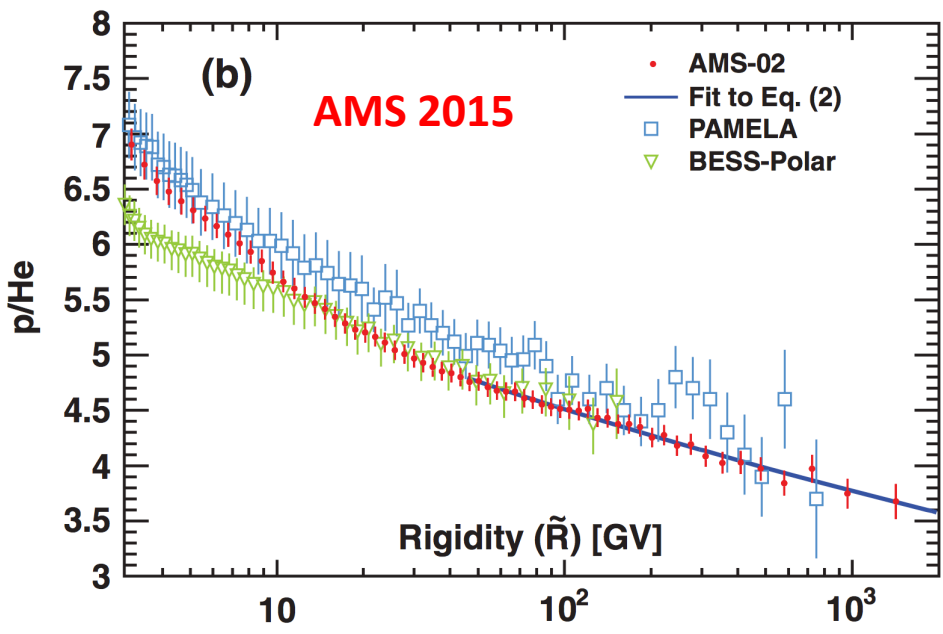


# AMS-100 - A Magnetic Spectrometer at Lagrange Point 2

- **AMS-02 is an excellent experiment on the Space Station. It is now operational since 7 years. We plan to operate it till the end of the lifetime of the ISS in 2024. Is is now time to think about the next step!**



## Recent AMS Results



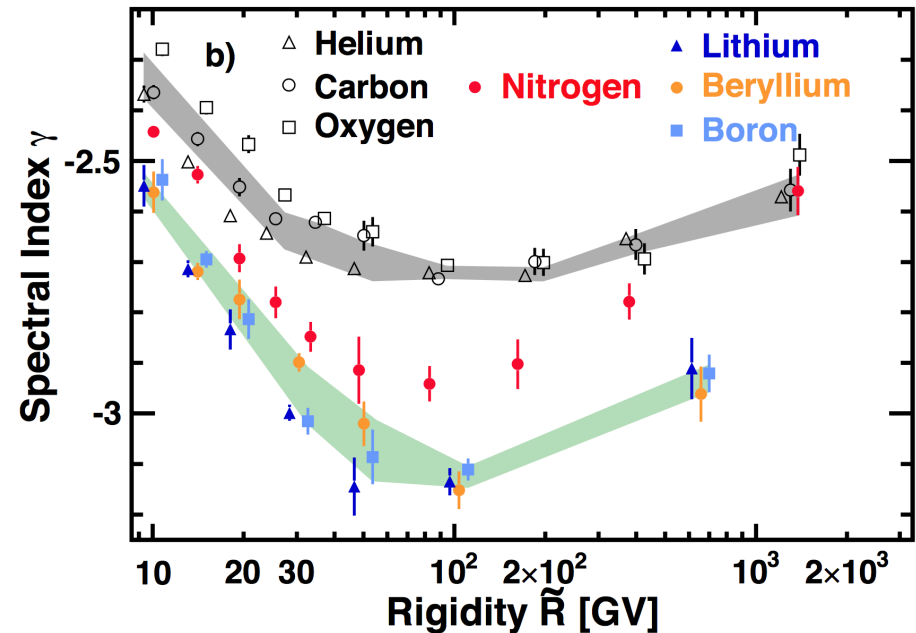
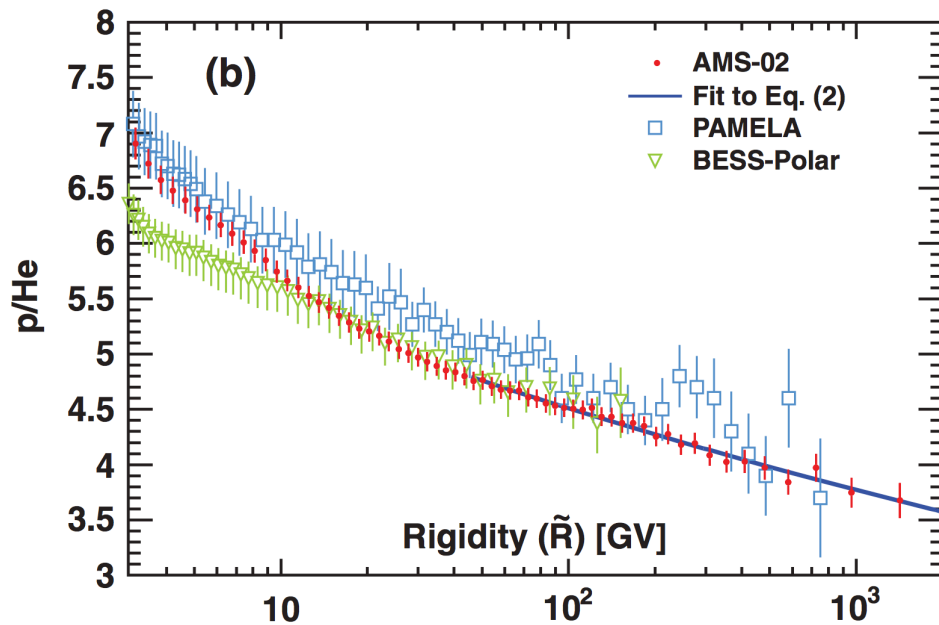
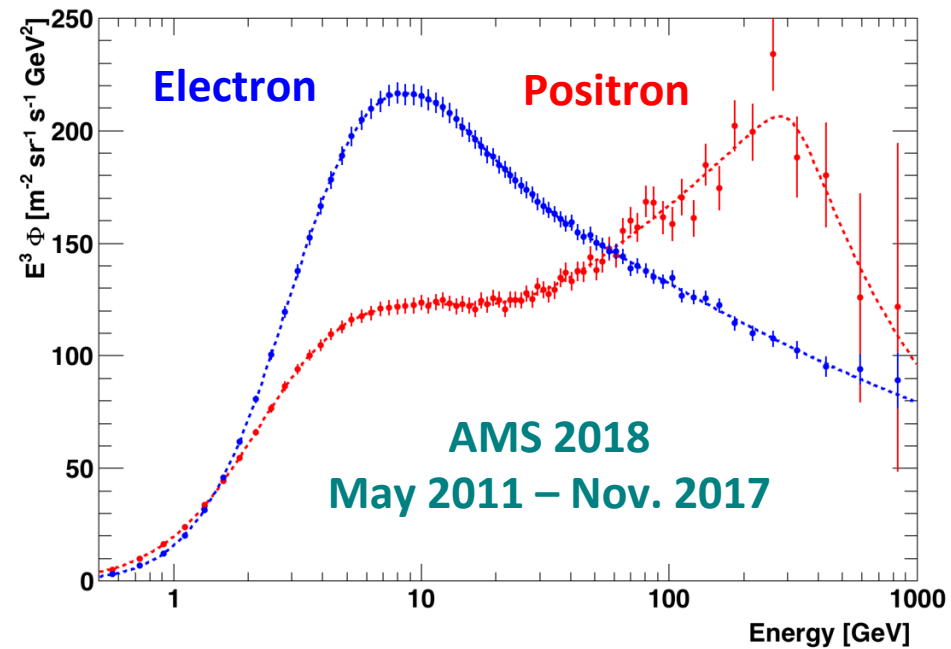
Discussion at this conference:

„We need to measure He/Proton to higher energies.“, Pasquale Blasi

„We need to measure photons in the energy range 300 GeV-3000 GeV with high precision.“, Ilias Cholis.

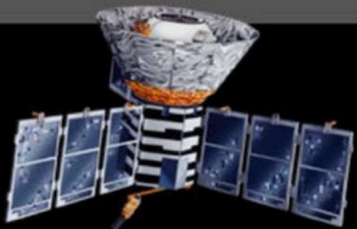
„To have the rigidity distribution for anti He would be great.“, Igor Moskalenko

As hosts we try to fulfill the wishes of our guests ...

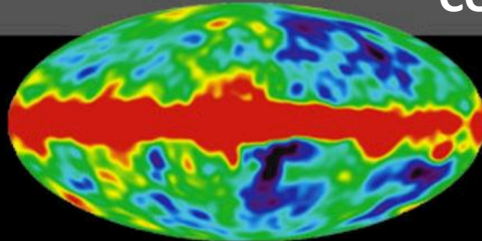


# AMS-100: A Magnetic Spectrometer at Lagrange Point 2

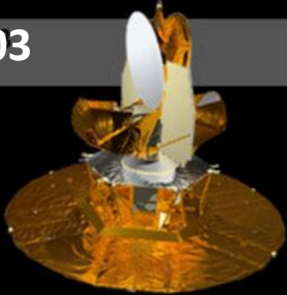
1992



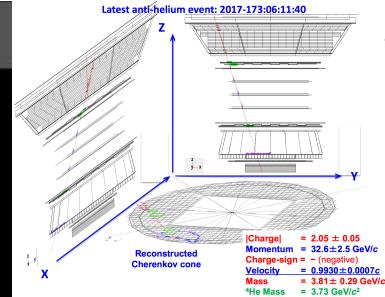
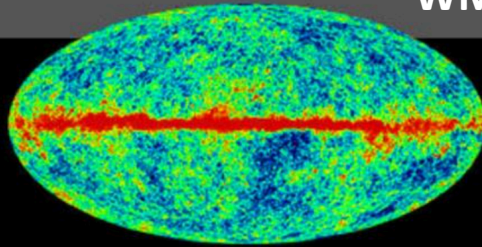
COBE



2003



WMAP



AMS-02  
2011



AMS-100  
2022

- Similar to COBE and WMAP or the Hubble- and the James Webb Space Telescope the only option for significant improvements compared to AMS-02 is a large magnetic spectrometer operated at Lagrange Point 2.
- A factor 10 in energy reach requires a factor 1000 in acceptance. AMS-02 has an acceptance of  $0.1 \text{ m}^2 \text{ sr}$  at high energies and weights 7 tons.

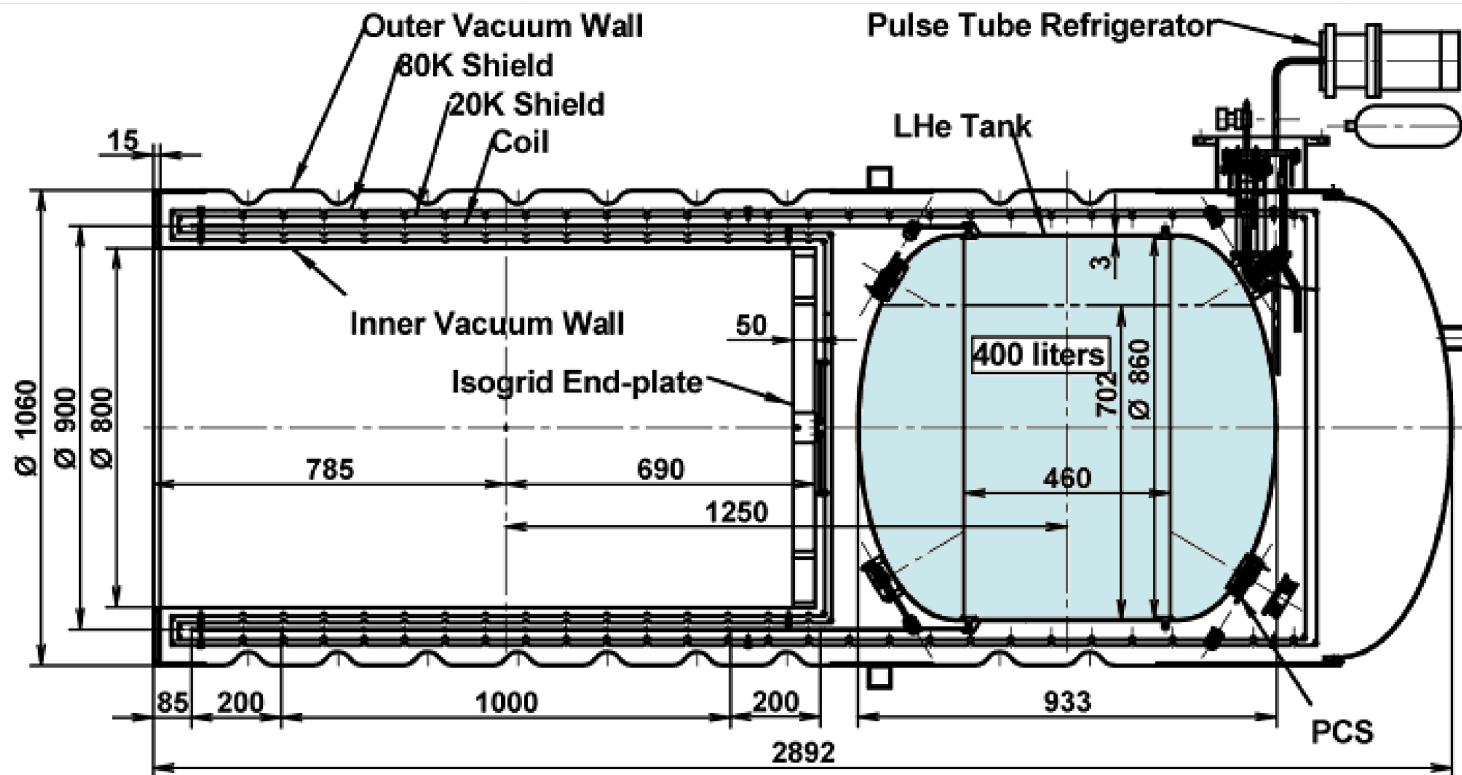
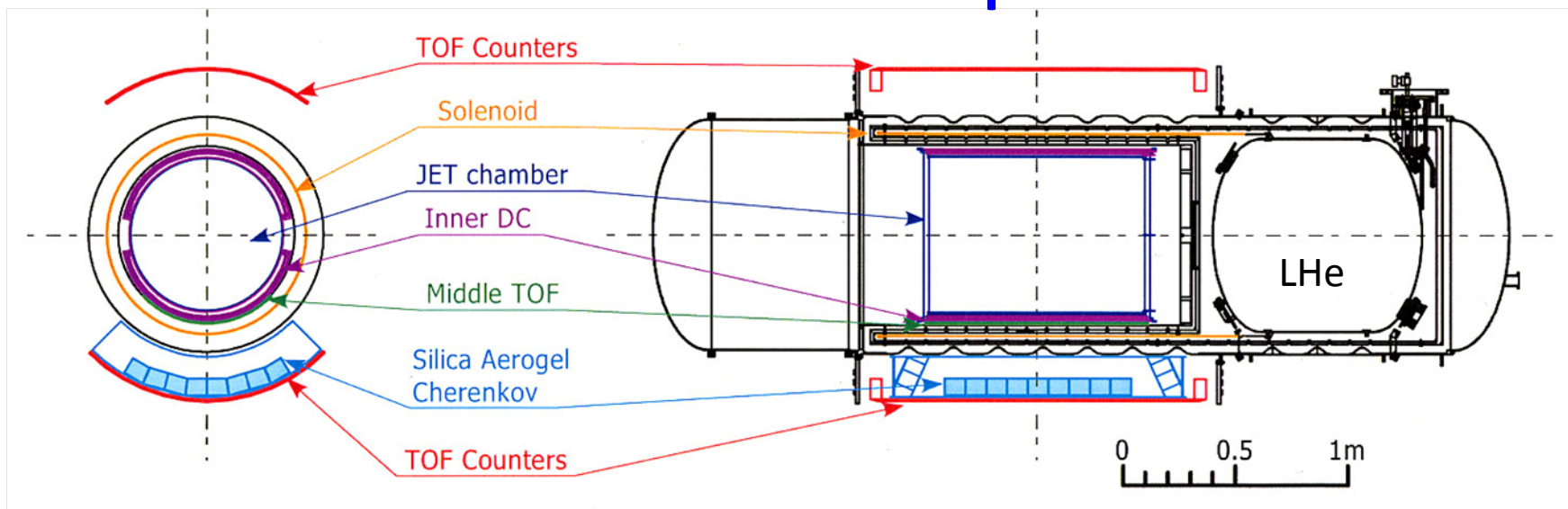
# BESS Polar – Balloon Experiment

December 13, 2004



**The concept of AMS-100 is inspired by the design of the successful BESS Polar Experiment.**

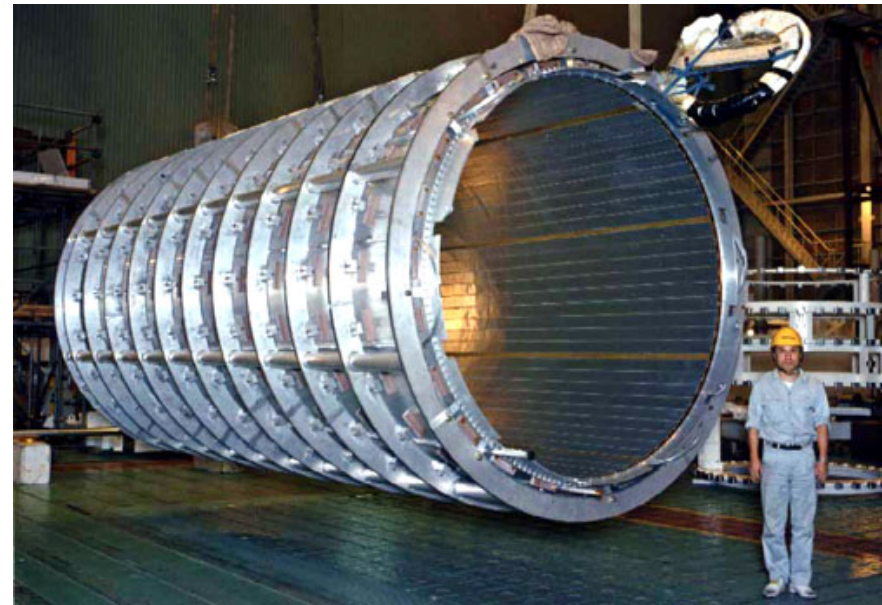
# BESS Polar – Balloon Experiment



# ATLAS Central Solenoid Magnet

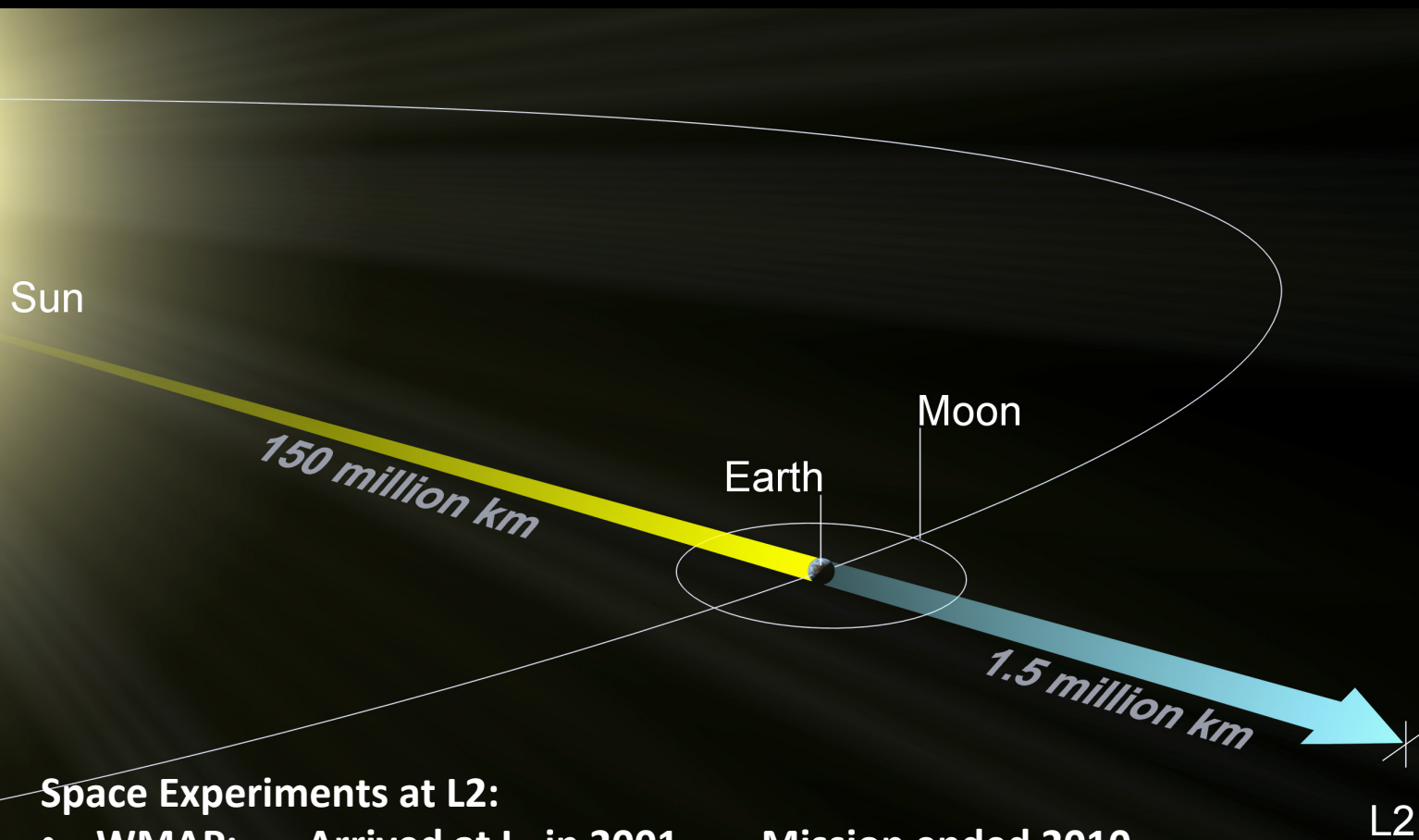
- only 0.66 radiation lengths X0 thick
- made from aluminium enforced Nb/Ti conductor
- operation temperature 4.5 K

- 5.3 long, 2.4 m diameter, 4.5 cm thick
- 5 tonne weight
- 2 tesla (T) magnetic field with a stored energy of 38 megajoules (MJ)
- 9 km of superconducting wire
- Nominal current: 7.73 kiloampere (kA)



# AMS-100: Operation at Lagrange Point 2

Due to earth magnetic field a cryogenic solenoid magnet can only be operated at L2.



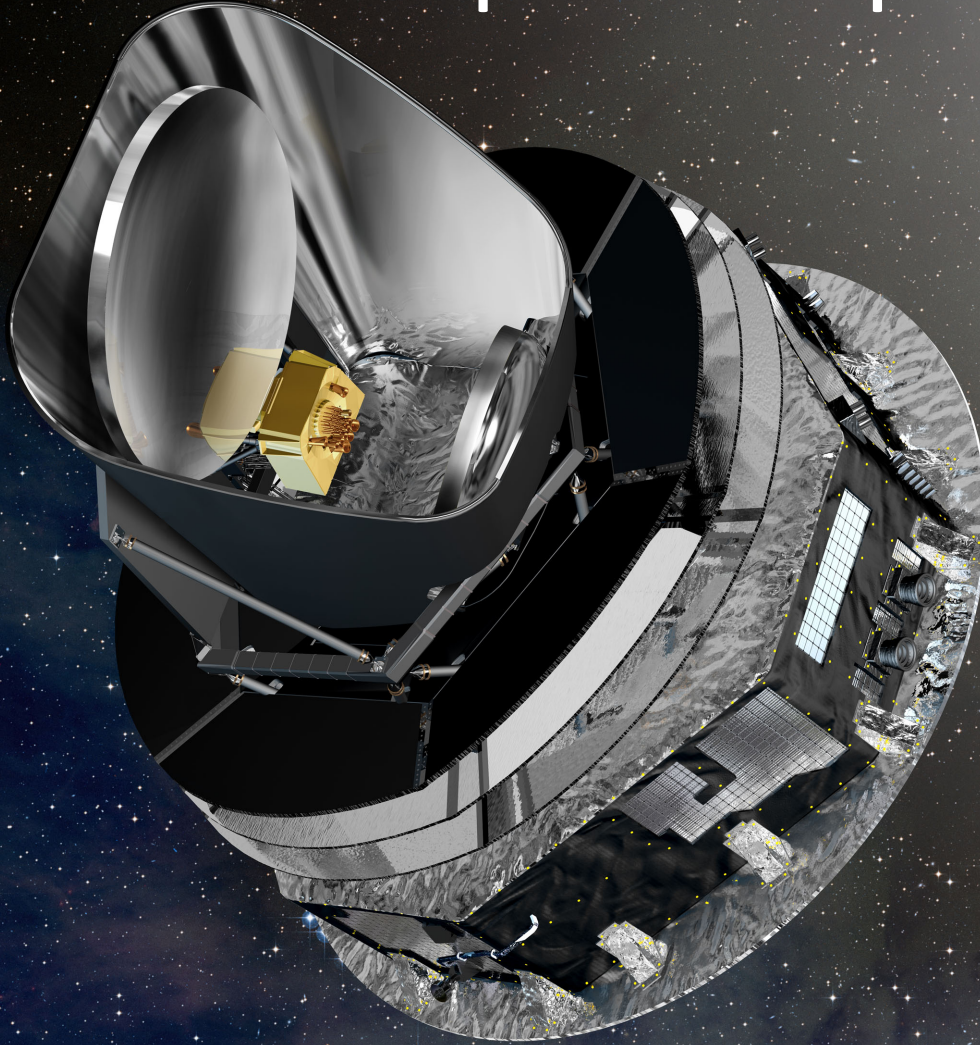
## Space Experiments at L2:

- **WMAP:** Arrived at L<sub>2</sub> in 2001. Mission ended 2010
- **Herschel:** Arrived at L<sub>2</sub> July 2009. Ceased operation on 29 April 2013
- **Planck:** Arrived at L<sub>2</sub> July 2009. Mission ended on 23 October 2013
- **Chang'e 2:** Arrived in August 2011 after completing a lunar mission before departing en route to asteroid 4179 Toutatis in April 2012.



# PLANCK

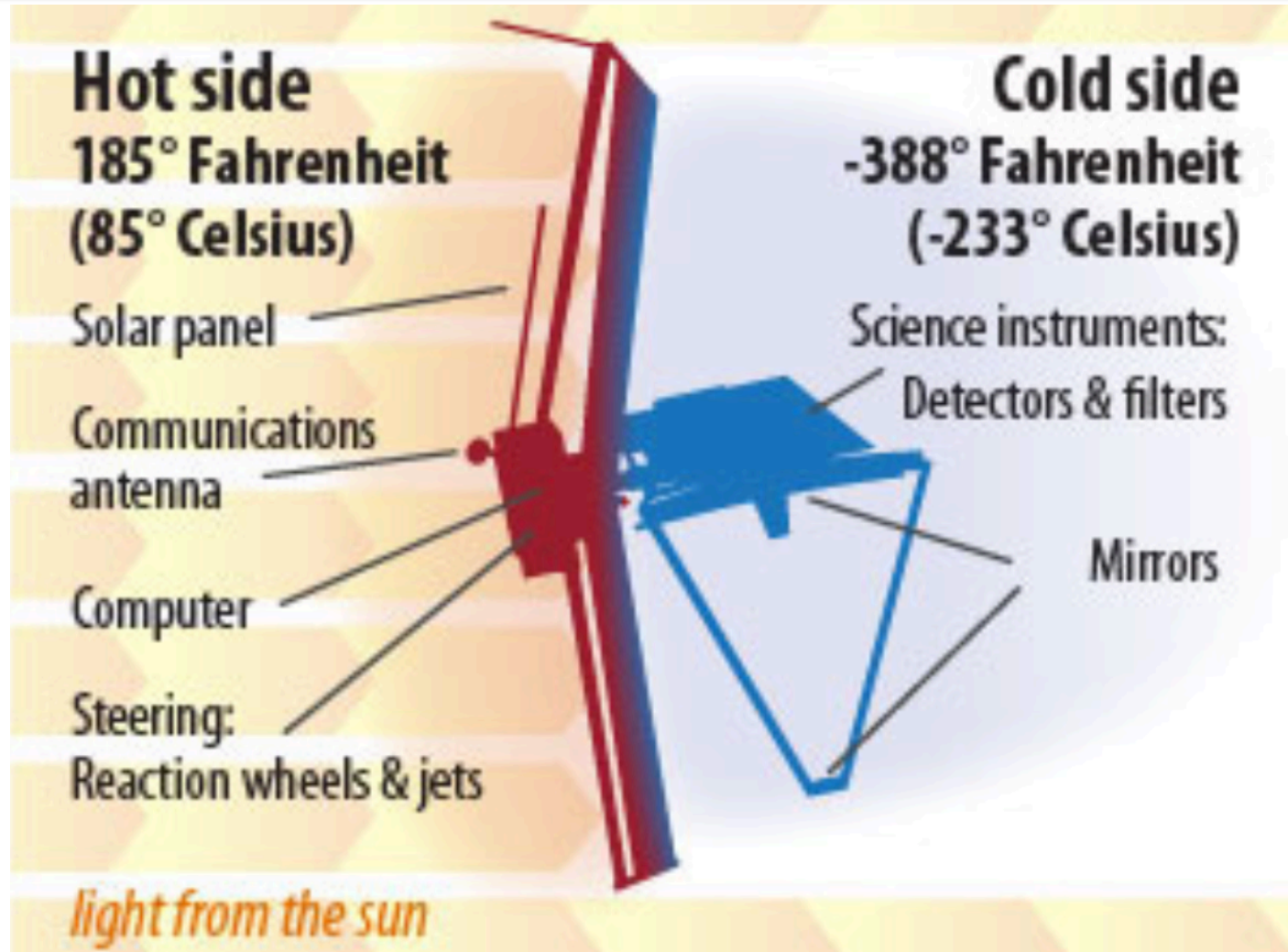
## Space Telescope



Planck's cooling systems allow it to maintain a temperature of  $-273.05\text{ }^{\circ}\text{C}$ . From August 2009, Planck was the coldest known object in space.

<b>Mission duration</b>	Planned: >15 months Final: 4 years, 5 months, 8 days
<b>Spacecraft properties</b>	
<b>Manufacturer</b>	Thales Alenia Space
<b>Launch mass</b>	1,950 kg (4,300 lb) <sup>[1]</sup>
<b>Payload mass</b>	205 kg (452 lb)
<b>Dimensions</b>	Body: 4.20 m × 4.22 m (13.8 ft × 13.8 ft)
<b>Start of mission</b>	
<b>Launch date</b>	14 May 2009, 13:12:02 UTC
<b>Rocket</b>	Ariane 5 ECA
<b>Launch site</b>	Guiana Space Centre, French Guiana
<b>Contractor</b>	Arianespace
<b>Entered service</b>	3 July 2009
<b>End of mission</b>	
<b>Disposal</b>	Decommissioned
<b>Deactivated</b>	23 October 2013, 12:10:27 UTC
<b>Orbital parameters</b>	
<b>Reference system</b>	L <sub>2</sub> point (1,500,000 km / 930,000 mi)
<b>Regime</b>	Lissajous

# James Webb Telescope - a space telescope at Lagrange Point 2

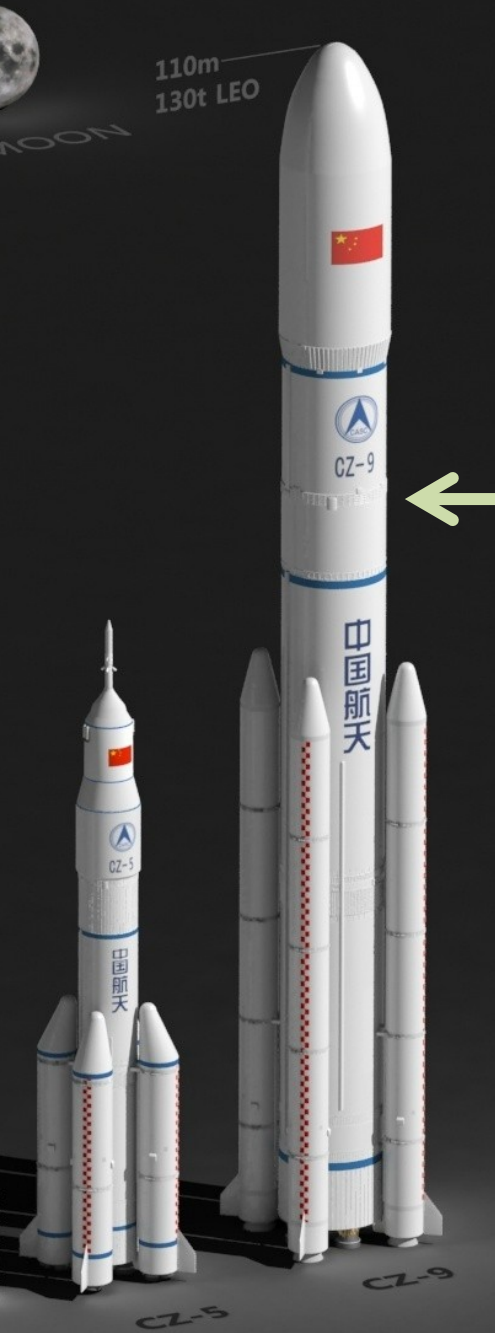


The science phase of the mission is expected to start in 2018  
and to last for 10.5 years.

This sunshield support structure is very strong, yet quite light, weighing only 63 kilograms (139 pounds), while supporting the sunshield itself, which weighs 700 kilograms (1,543 pounds).

21.197 m × 14.162 m





## Current and upcoming rockets

Name	LEO [kg]	other [kg]	First flight	
Ariane 5	21,000	10,730 GTO	2002	ESA
Falcon Heavy	63,800	26,700 GTO	2018	SpaceX
Long March 5	25,000	8,000 TLI	2016	CALT
Long March 9	130,000	50,000 TLI	2025	CALT
SLS Block 1B	105,000	39,100 TLI	2022	NASA
SLS Block 2	130,000	45,000 TLI	2025	NASA

Operational

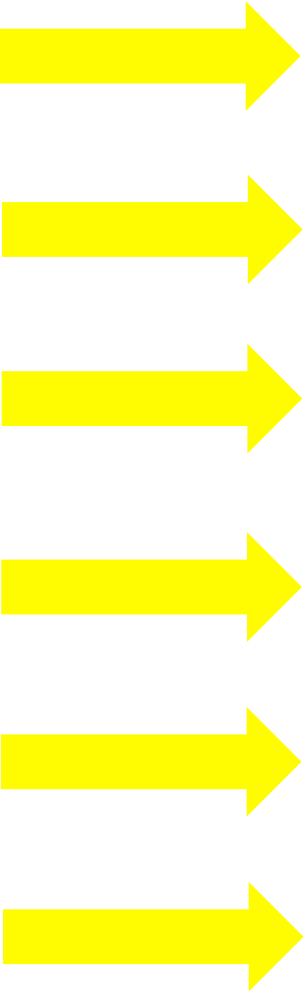
Under development

LEO: Low Earth orbit

GTO: Geostationary transfer orbit

TLI: Trans-lunar injection

Sun



Sun Shield

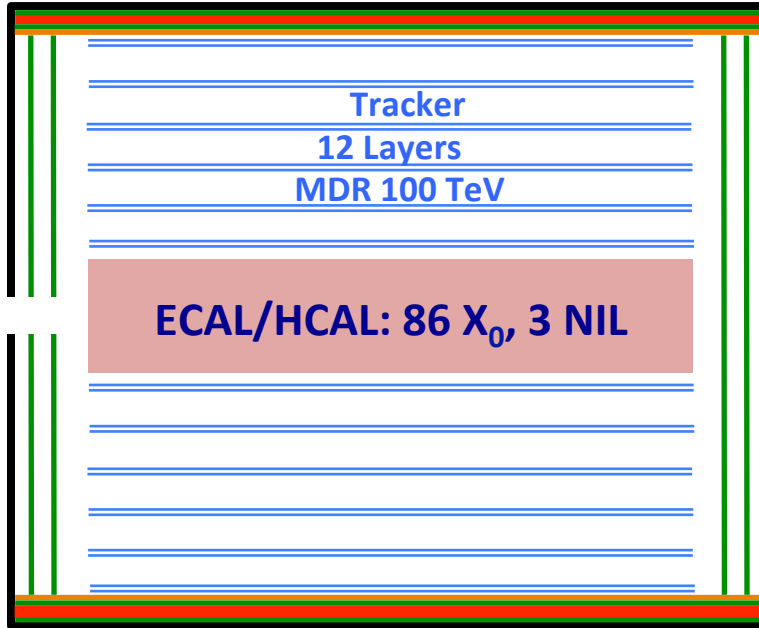
4 m



Space-  
craft  
Bus

AMS-100 is rotational symmetric around the z-Axis.

Thin Solenoid 1 Tesla,  
3.5 m inner diameter, Length 5 m

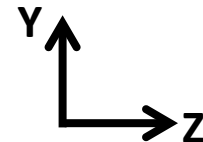


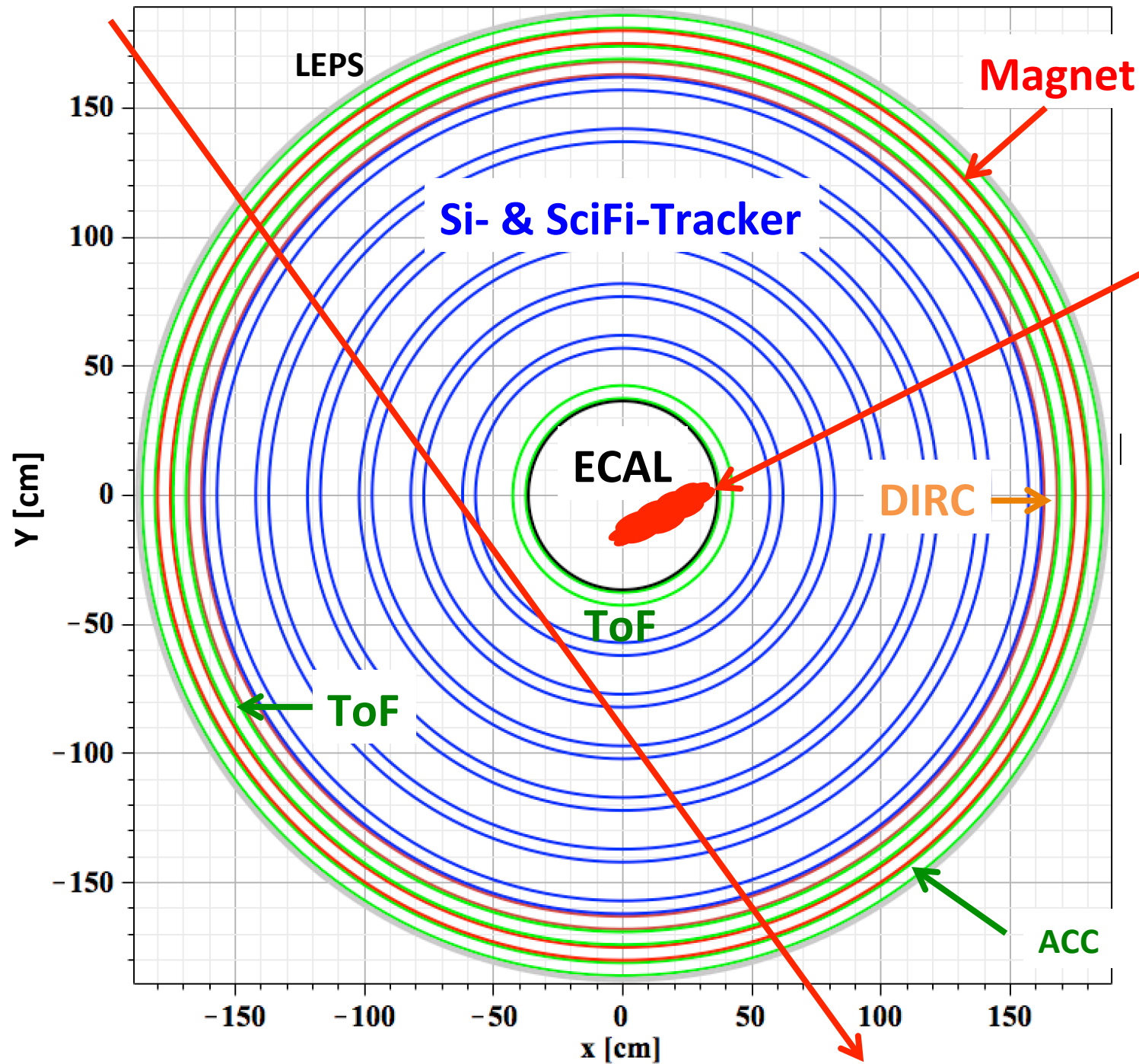
5 m



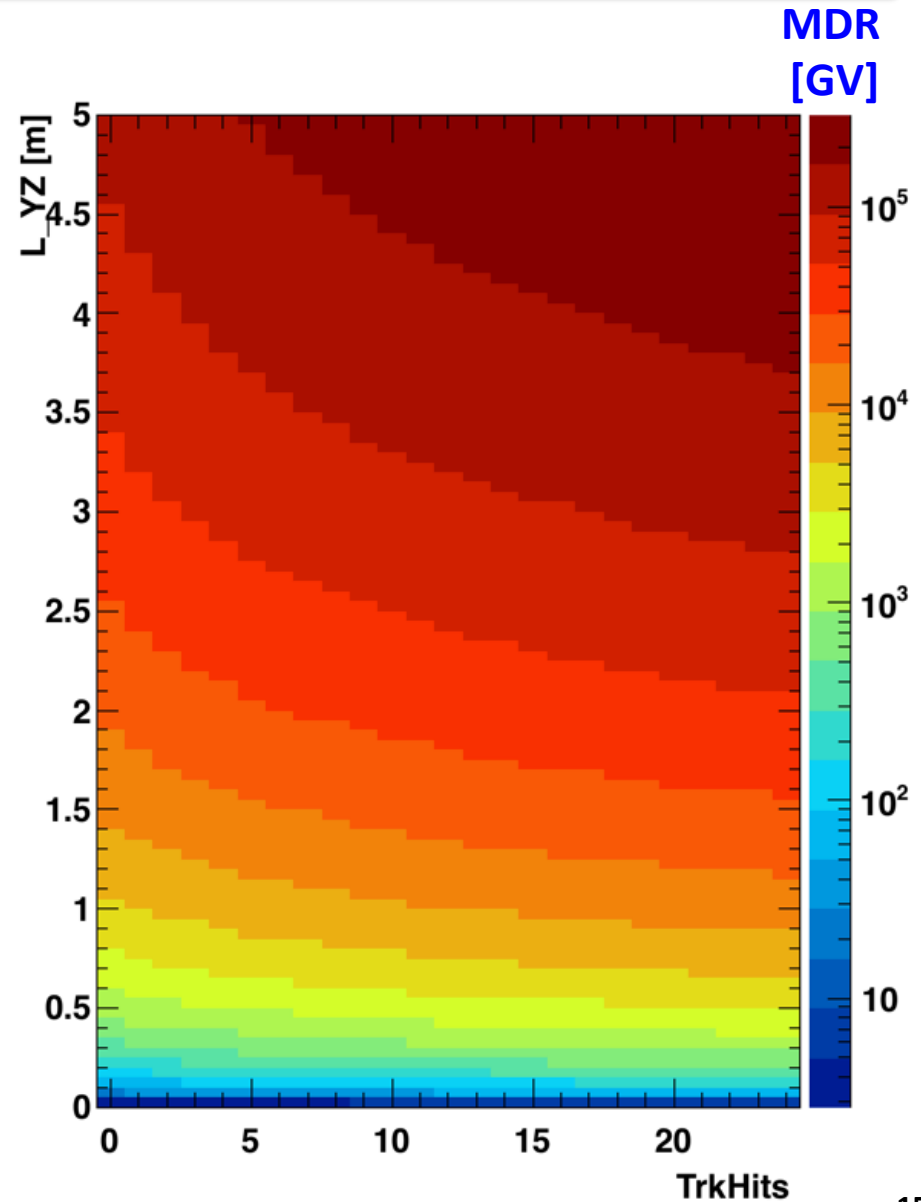
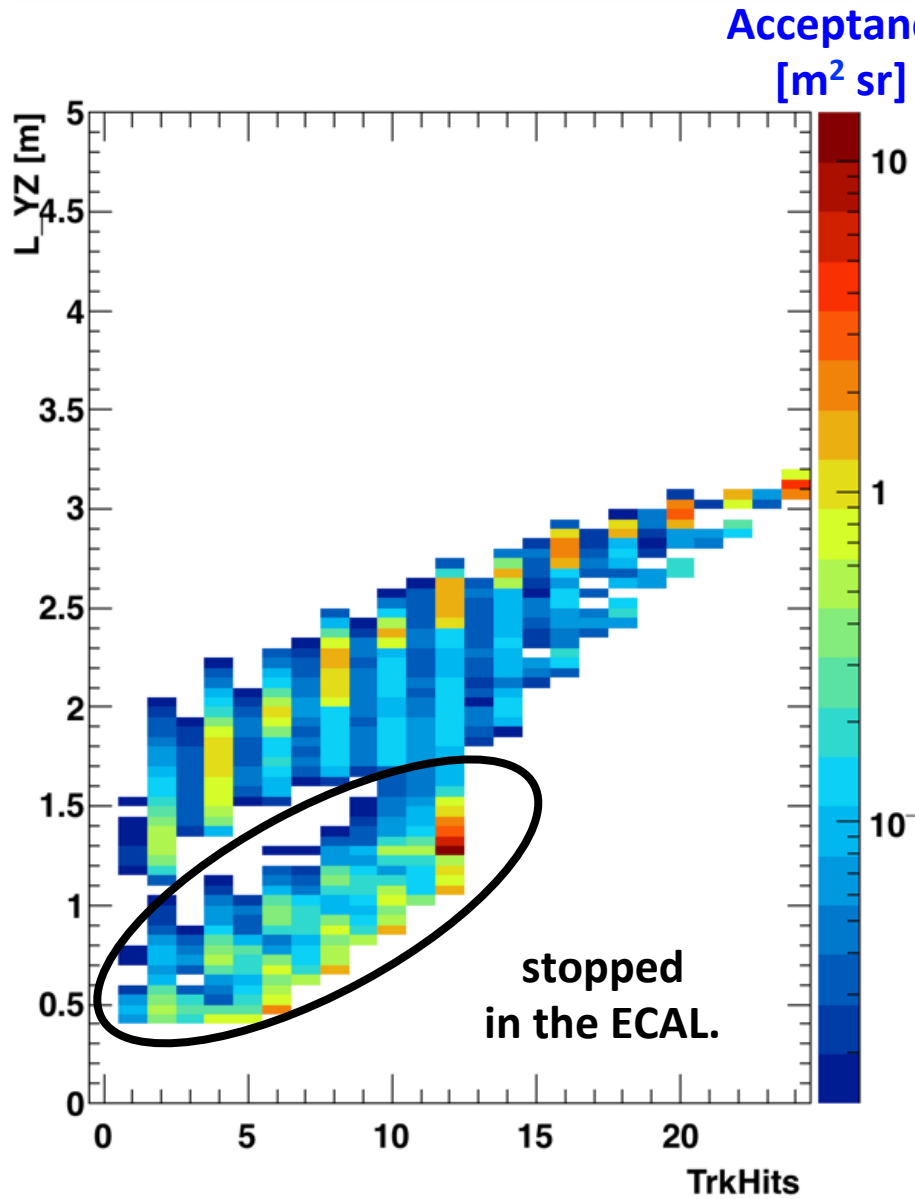
He –  
Vessel  
20 m<sup>3</sup>

Spacecraft Bus:  
(Electrical Power, Attitude  
Control, Communication,  
Command & Data Handling,  
Propulsion, Thermal Control)



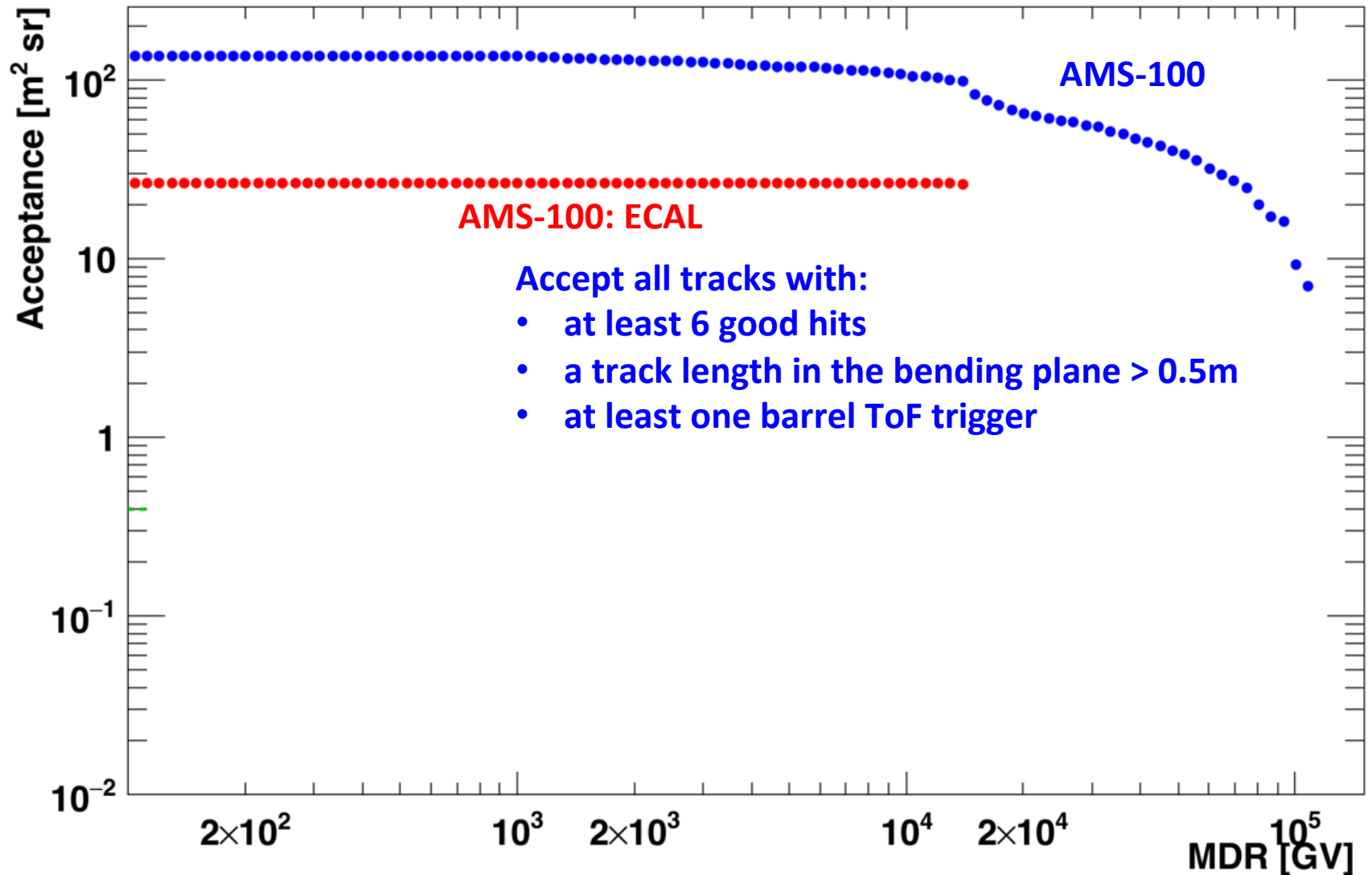


# Acceptance & MDR



# Acceptance

At 10 TeV: Tracker: 81 [m<sup>2</sup> sr]    ECAL: 26 [m<sup>2</sup> sr]    Total: 107 [m<sup>2</sup> sr]  
At 100 TeV: Tracker: 10 [m<sup>2</sup> sr]    ECAL: 0 [m<sup>2</sup> sr]    Total: 10 [m<sup>2</sup> sr]





# AMS-100: A Magnetic Spectrometer at LP-2

Weight: 43 t

Readout-Channels:  $4 \cdot 10^6$

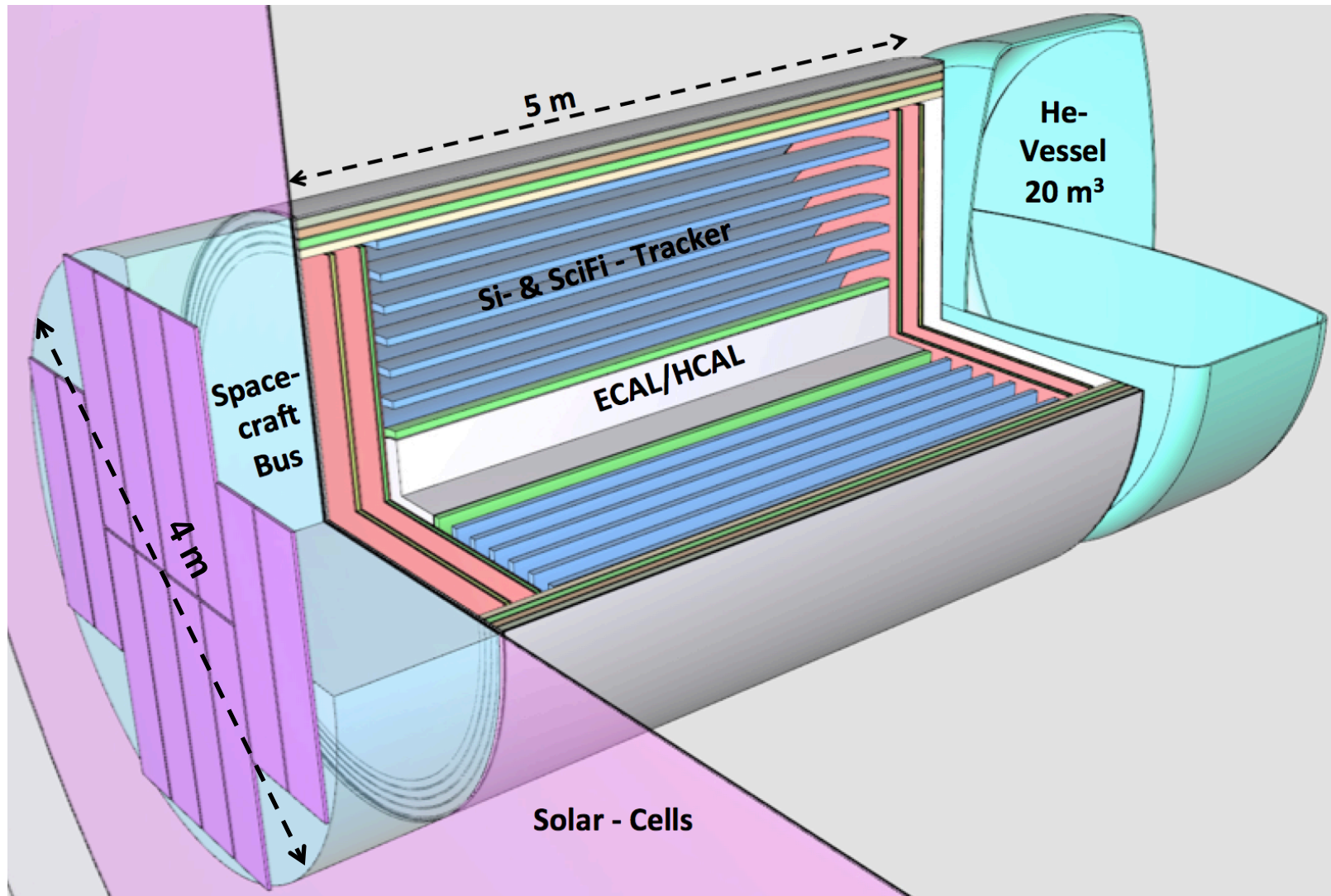
Power: 40 kW

Trigger Rate: 1 MHz

MDR: 100 TeV

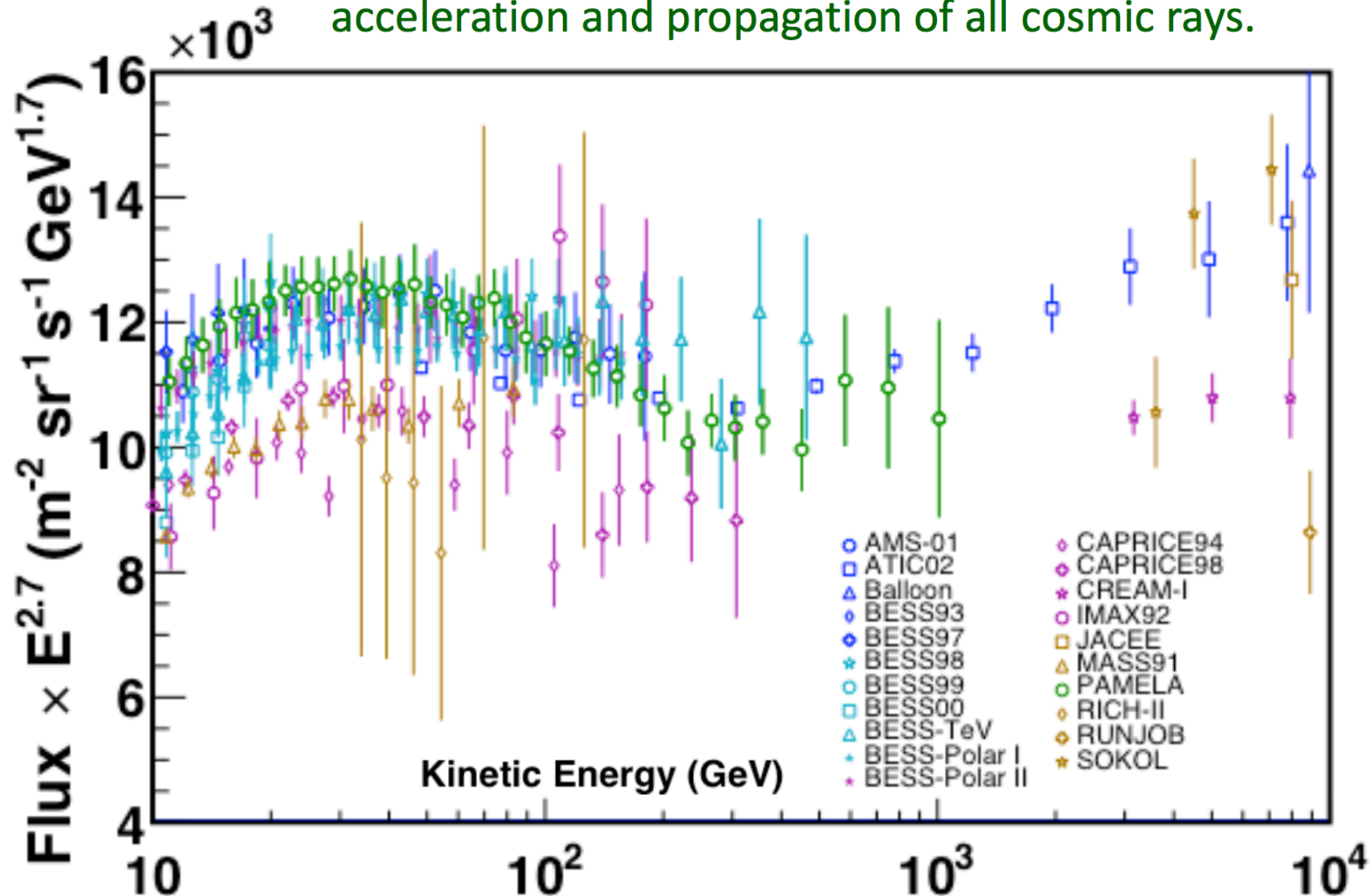
Acceptance:  $100 \text{ m}^2 \text{ sr}$

B-Field: 1 Tesla

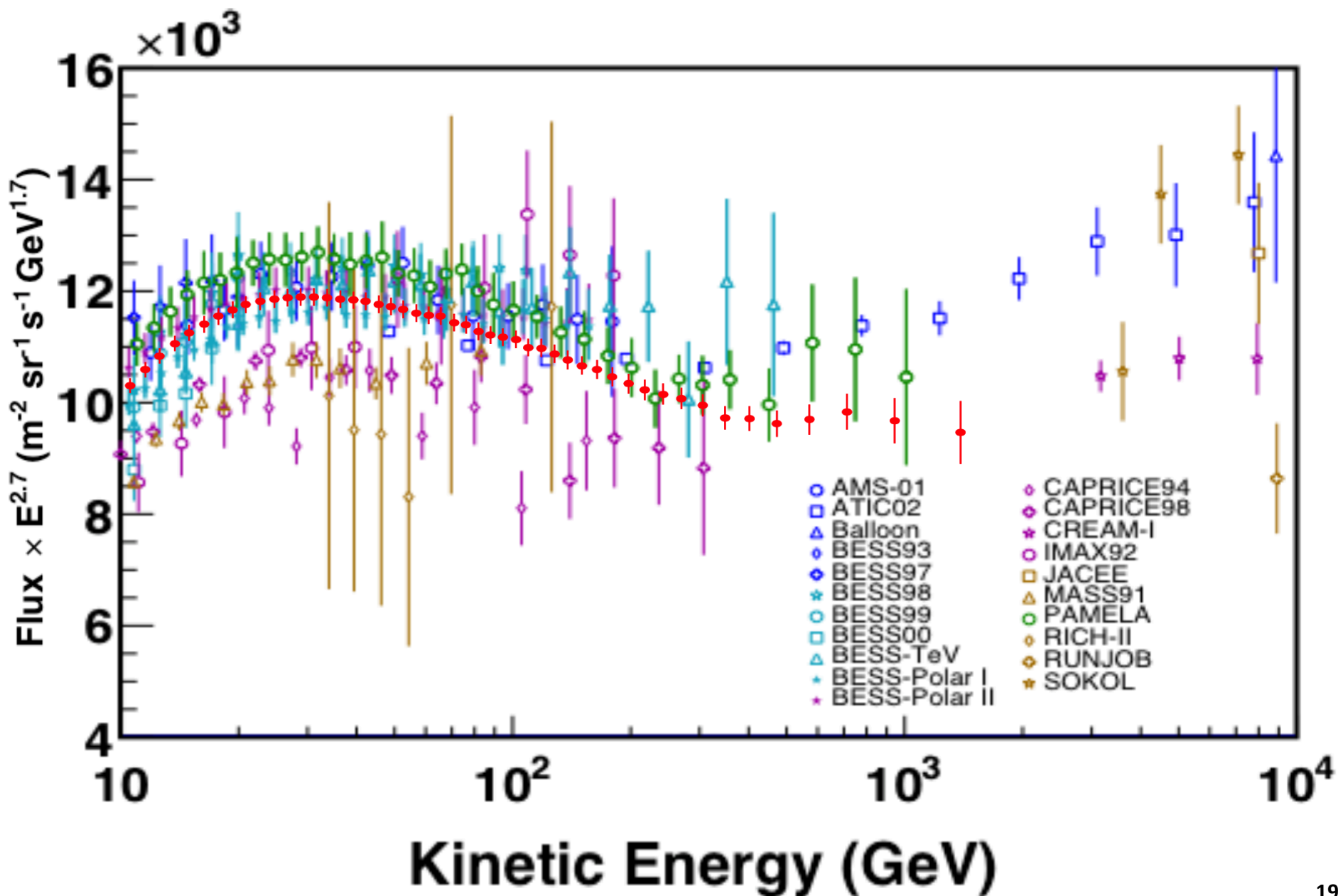


# Measurements of proton spectrum before AMS

1. Protons are the most abundant charged cosmic rays.
2. These were the best data over the last hundred years.
3. Nonetheless, the data have large errors and are inconsistent.
4. These data limit the understanding of the production, acceleration and propagation of all cosmic rays.

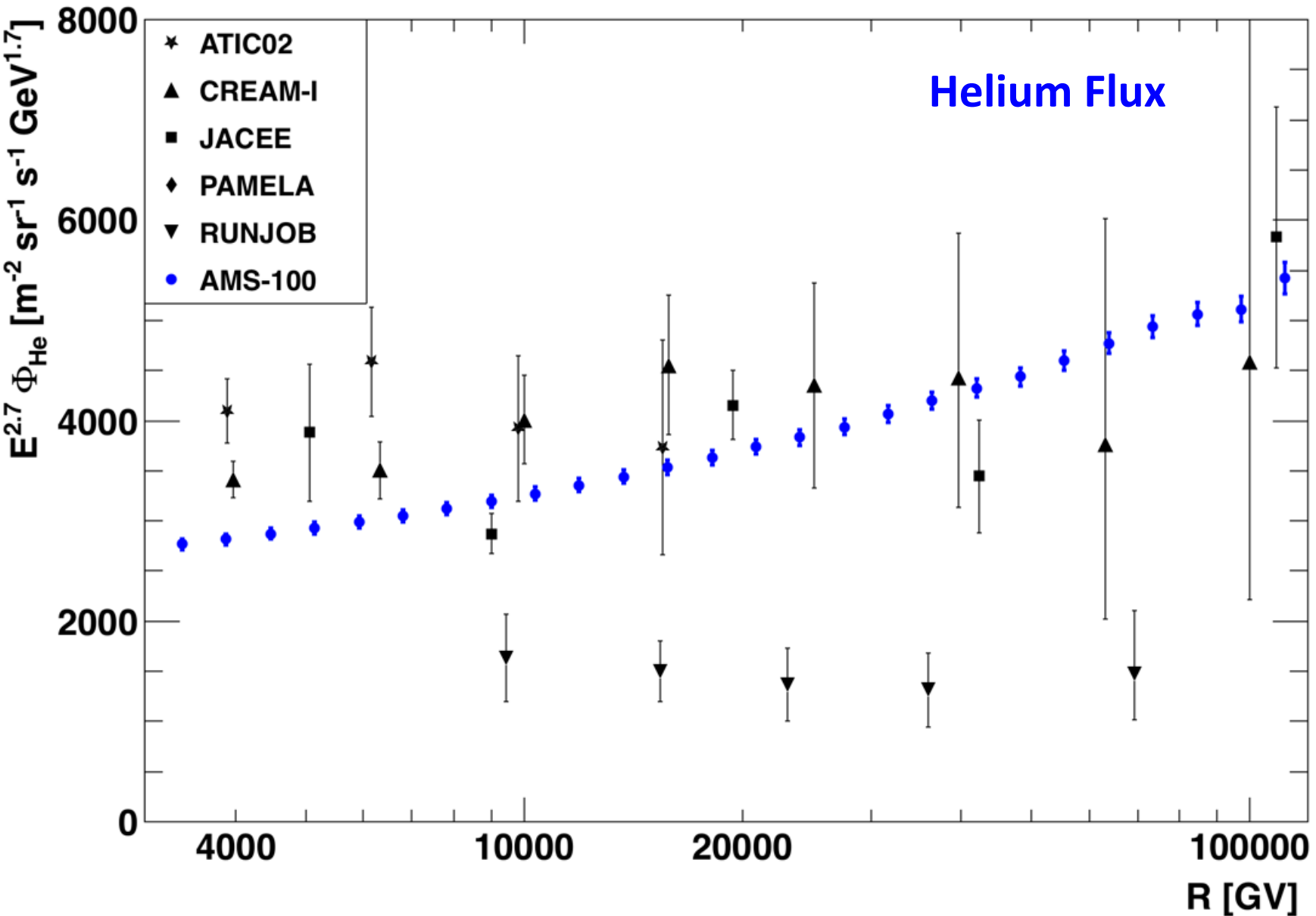


# AMS Measurement of the proton spectrum



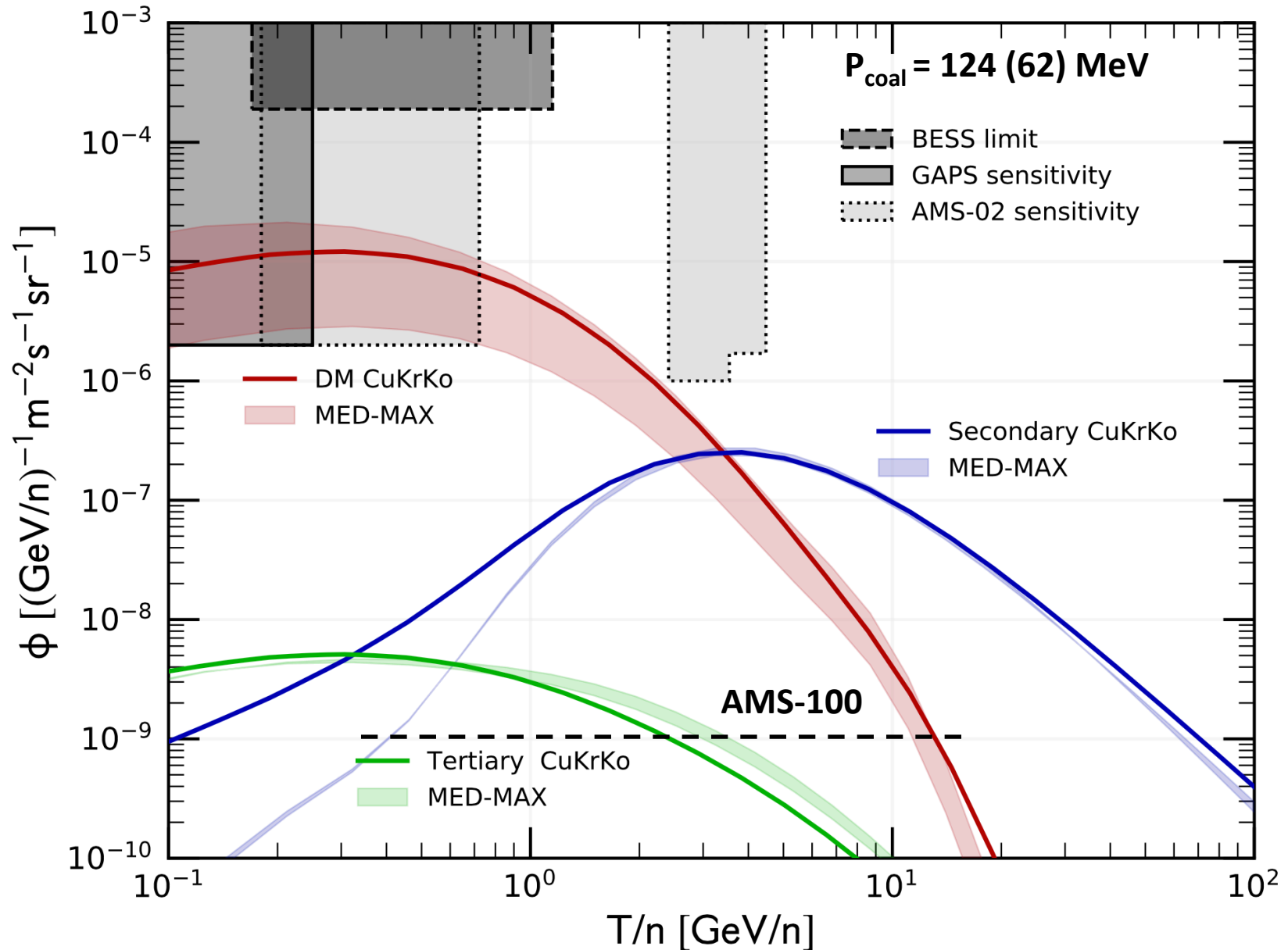


We don't know what we would find in this energy region  
with an instrument with 1% accuracy and an MDR of 100 TV.

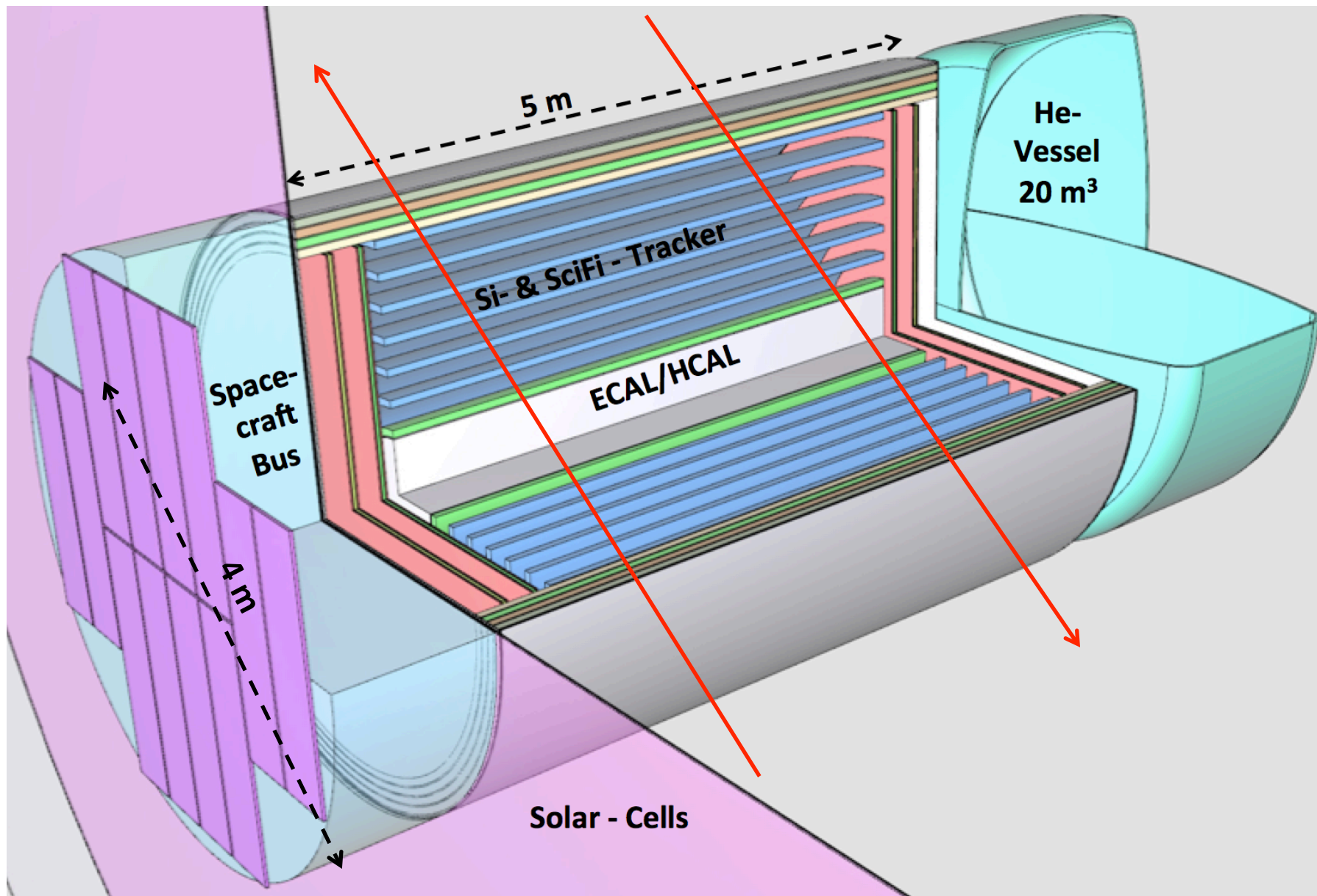


# Anti-Deuteron

F. Donato, Fornengo, Korsmeier, 1711.08465 subm. PRD



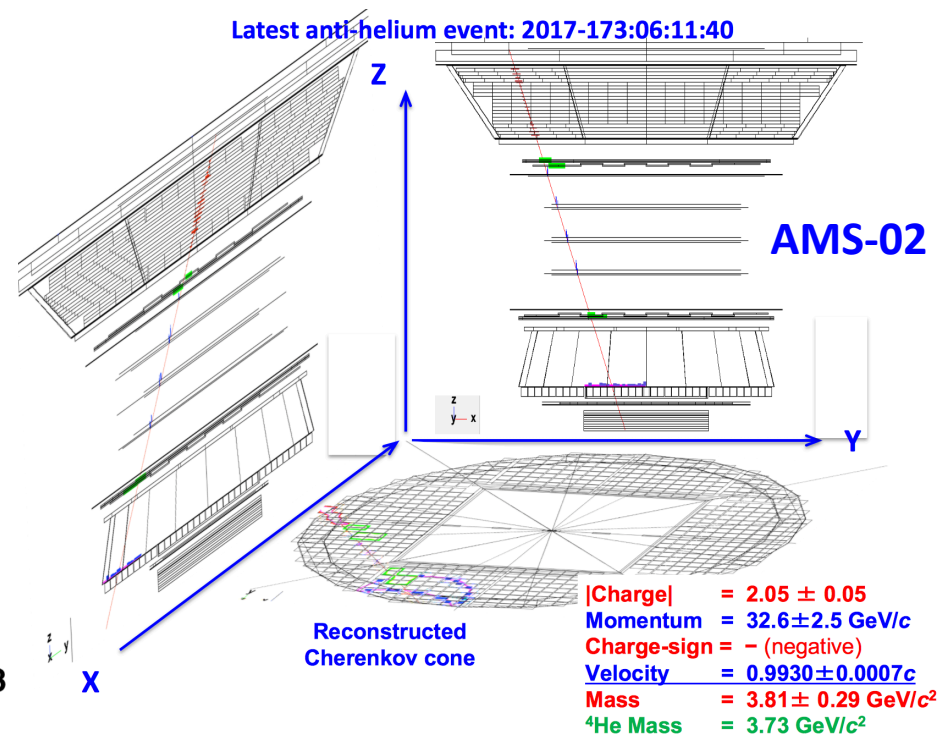
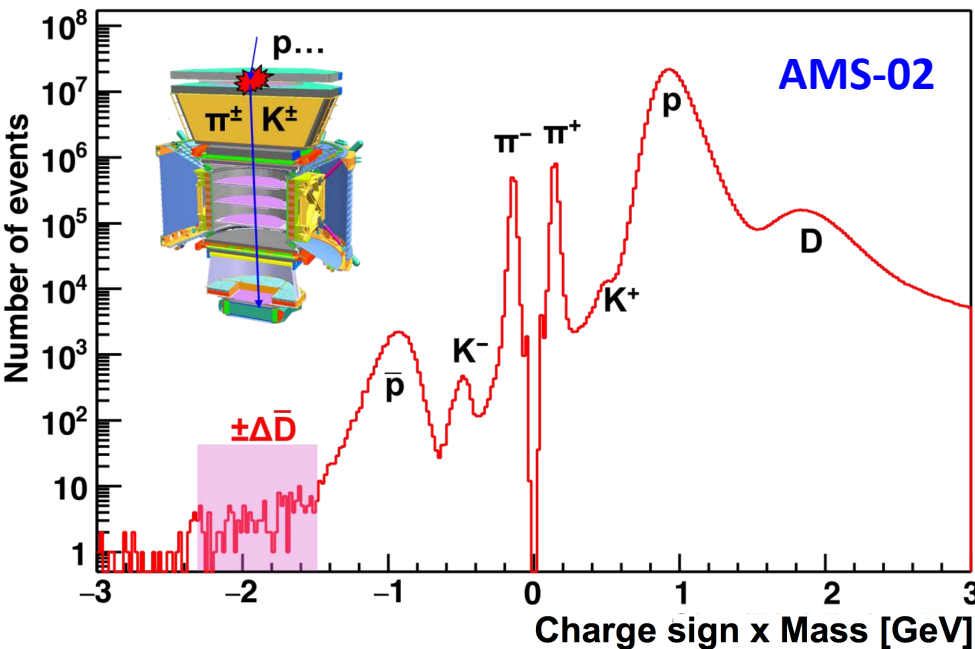
Due to the rotational symmetry of the instrument one would expect the same number of anti-He particles going upwards or going downwards.  
This is equivalent to a change of the polarity of the magnetic field.



# AMS-100: Charged Cosmic Rays

- Proton and Helium Spectra in the rigidity range  $R=500 \text{ GV} - 100 \text{ TV}$
- Carbon and Oxygen Spectra in the rigidity range  $R=100 \text{ GV} - 50 \text{ TV}$ .
- Lithium, Beryllium and Boron Spectra in the rigidity range  $R=100 \text{ GV}-30 \text{ TV}$ .
- Electron and Positron Spectra in the energy range  $E=100 \text{ GeV} - 10 \text{ TeV}$ .
- Anti-Proton Spectrum in the rigidity range  $R=100 \text{ GV} - 10 \text{ TV}$ .

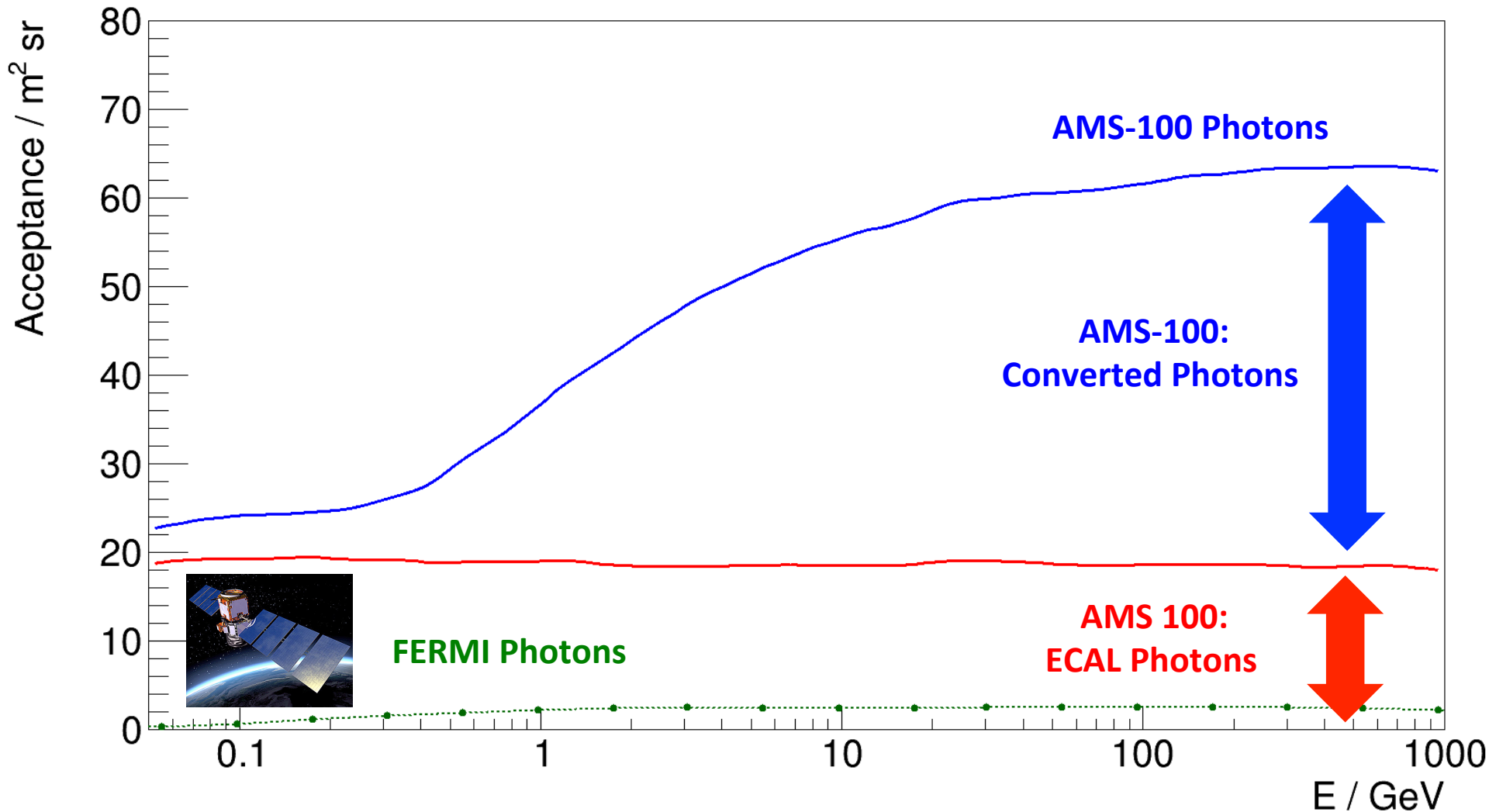
If AMS-02 identifies Anti-Deuterium, Anti-Helium 3 and Anti-Helium 4 in primary cosmic rays, AMS-100 would be able to measure the spectra with  $>1000$  Events for each species.





# AMS-100 will monitor the whole sky continuously.

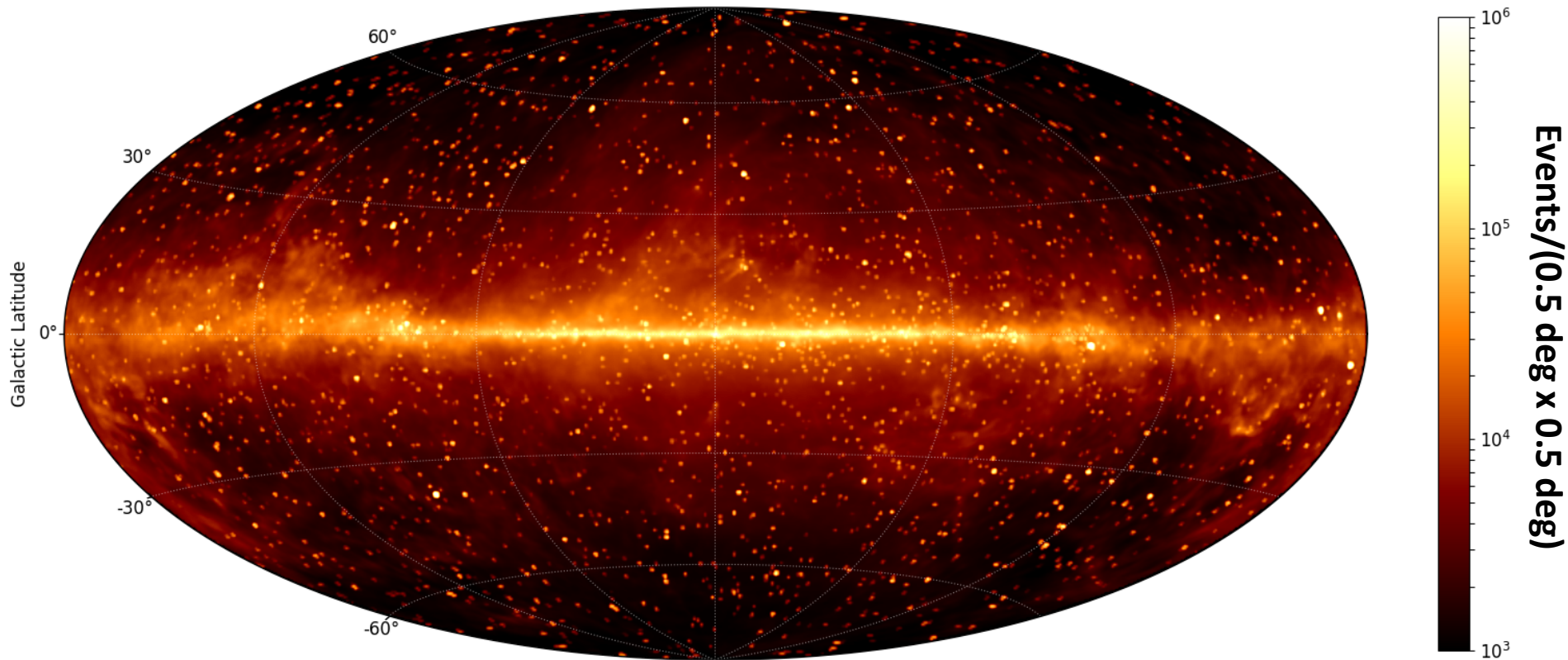
The acceptance for photons is up to  $65 \text{ m}^2 \text{ sr}$  compared to FERMI  $2.5 \text{ m}^2 \text{ sr}$ .



The angular Resolution for converted photons is  $0.005 \text{ mm}/3.5\text{m} = 1 \cdot 10^{-4} \text{ deg}$  and will be  **$\sim 1,000$  better** than the FERMI resolution.

**The 3. FERMI catalog includes 3033 sources above 4 sigma significance.**

**AMS-100: Expected counts for 5 years for E=50 MeV – 1 TeV**



- **For every source in the 3. FERMI catalog we expect at least 1000 events in AMS-100.**
- **We expect to see 10000 new sources in AMS-100.**

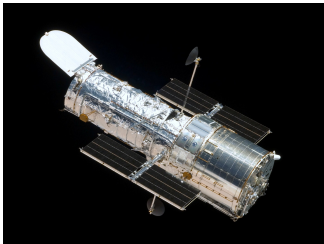
# AMS-100: A Magnetic Spectrometer at LP-2

- The scale of this project is the scale of a LEP Experiment ⇔ inner part of (ATLAS,CMS)  
⇔ 250 Million (€, \$, CHF) but for space application i.e. 500 Million - 1000 Million (€, \$, CHF).
- It requires a collaboration of 500 physicists, it is time to start the R&D now.

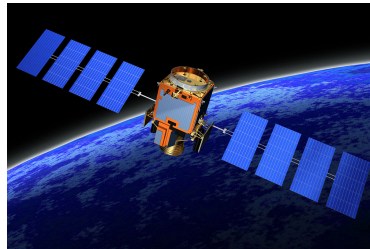
# AMS-100: A Magnetic Spectrometer at LP-2

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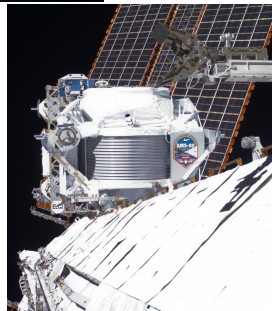
1990  
Hubble



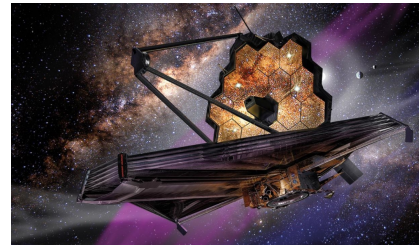
2008  
FERMI



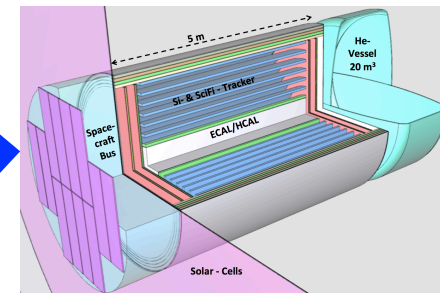
2011  
AMS-02



2020  
James Webb



2030  
AMS-100



# AMS-100: A Magnetic Spectrometer at LP-2

- A large scale superconducting magnet in space has large implications for human space exploration.
- To get a large scale superconducting magnet into space was even for S. Ting a challenge.
- We need a precursor flight with AMS-10, i.e. Acceptance  $10 \text{ m}^2 \text{ sr}$ , MDR 10 TeV, to LP-2 in 2025 to prove the technical concept.

# AMS-100: A Magnetic Spectrometer at LP-2

- A large scale superconducting magnet in space has large implications for human space exploration.
- To get a large scale superconducting magnet into space was even for S. Ting a challenge.
- We need a precursor flight with AMS-10, i.e. Acceptance  $10 \text{ m}^2 \text{ sr}$ , MDR 10 TeV, to LP-2 in 2025 to prove the technical concept.



**Second Nobel Prize for Samuel C. C. Ting from MIT for the discovery of anti-matter in space.**

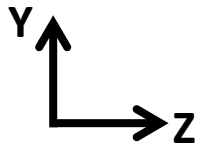
**AMS-10 first data confirmed the AMS-2 discoveries of anti-helium in space.**

**AMS-10 is the first space experiment with a large superconducting magnet and was launched in 2025.**

# Backup Slides

## Magnet

- Thin superconducting solenoid,  $B=1$  Tesla,  $T=4.5$  Kelvin made from aluminium enforced Nb/Ti conductor
- Inner diameter 3.5 m, Length 5 m, Material 0.48 X0
- Weight 7,400 kg (11,500 kg extrapolated from BESS, 3,500 kg extrapolated from ATLAS)
- 20 m<sup>3</sup> liquid He (2400 kg) should cool the magnet for >6 years. Vessel weight estimated to be 1700 kg.





```
> n := 140/Unit(m) : # turns/m
Ix := 7.73*Unit(kA) : # from ATLAS solenoid
Magnet_Wire_Length := simplify(n*Magnet_LenY*2.*Pi*rMagnet_in);
Magnet_Wire_Length := 7696.902002 [[m]]
```

(3.1)

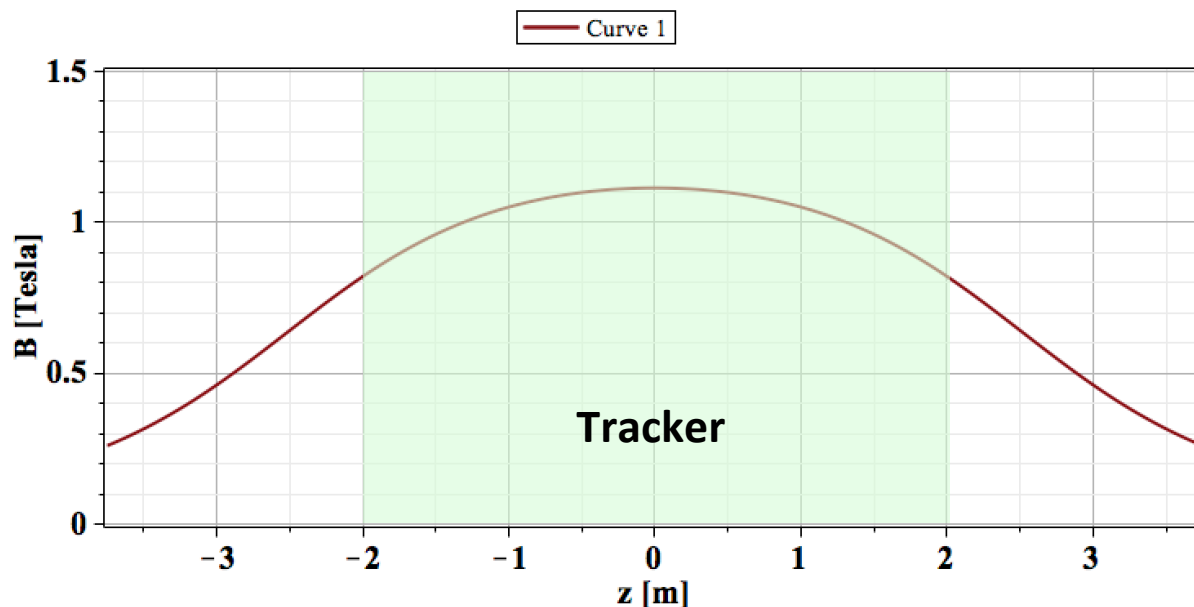
```
> x1 := -0.5*Magnet_LenY: x2 := 0.5*Magnet_LenY:
R := rMagnet_in:
mu_0 := evalf(Constant(mu[0], units)):
fB := x->simplify(mu_0*n*Ix/2*((x-x1)/sqrt((x-x1)^2+R^2)-(x-x2)/sqrt((x-x2)^2+R^2)));
fB := x->simplify(1/2 mu_0 n Ix ( (x-x1)/sqrt((x-x1)^2+R^2) - (x-x2)/sqrt((x-x2)^2+R^2) ))
```

(3.2)

```
> B_TrackerCenter := fB(0*Unit(m));
B_TrackerEdge := fB(0.5*SiTrk_LenY);
B_Mean := int(fB(x*OneM), x=-0.5*SiTrk_LenY/OneM..0.5*SiTrk_LenY/OneM)*OneM/SiTrk_LenY;
B_TrackerCenter := 1.114100220 [[T]]
B_TrackerEdge := 0.8205328942 [[T]]
B_Mean := 1.022762664 [[T]]
```

(3.3)

```
> plot(fB(x*OneM)/OneT, x=simplify(1.5*x1/OneM)..simplify(1.5*x2/OneM), gridlines=true, y=0..1.5,
labels=["z [m]", "B [Tesla]"]);
```



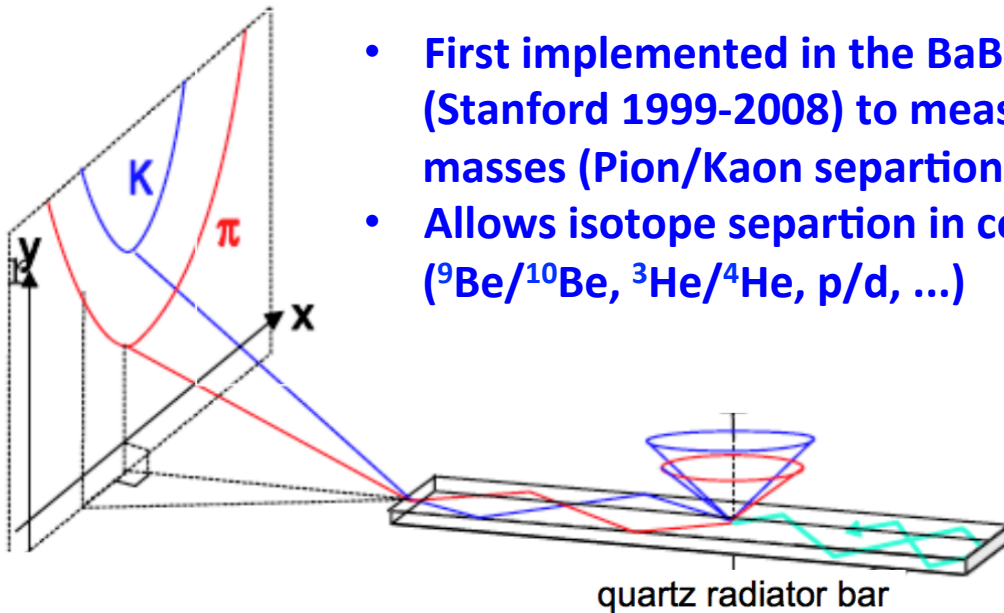


The diagram shows a cross-section of a detector assembly. At the top and bottom, there are several layers of material, represented by horizontal bands of different colors: blue, light blue, red, green, and orange. In the center, there is a large white rectangular region labeled 'Time of Flight (ToF)'. On the left and right sides of this central region, there are vertical structures. Each side has two vertical columns of material, one green and one blue, with a thin grey line between them. A coordinate system is shown in the bottom-left corner of the central region, with a vertical arrow labeled 'y' pointing upwards and a horizontal arrow labeled 'z' pointing to the right.

## Time of Flight (ToF)

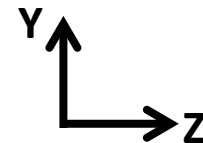
- Provides the Trigger and measures the flight time
- Scintillator Tiles with wavelength shifting fibers and SiPMT readout.
- Two layers of 10 mm thick tiles on a CF-honeycomb support structure.
- Time resolution: < 120 ps

## DIRC: Detection of Internally Reflected Cherenkov light

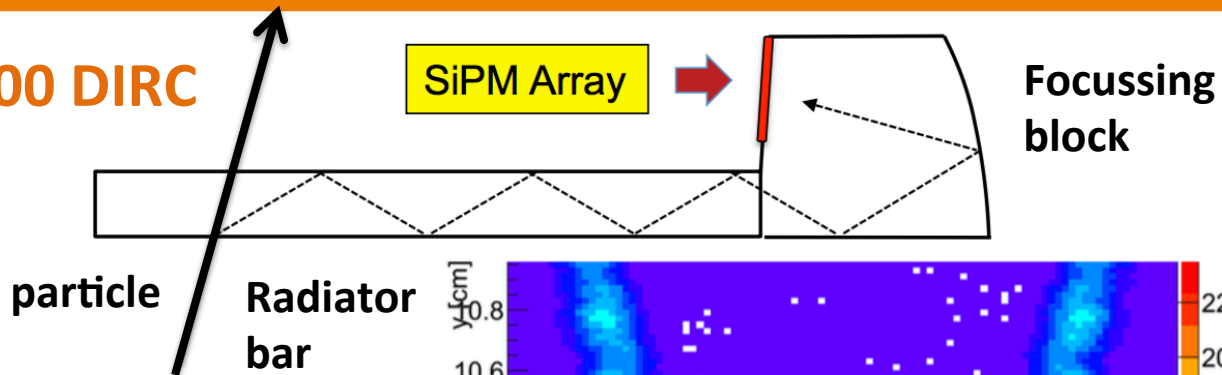


- First implemented in the BaBar Experiment (Stanford 1999-2008) to measure particle masses (Pion/Kaon separation).
- Allows isotope separation in cosmic rays ( $^9\text{Be}/^{10}\text{Be}$ ,  $^3\text{He}/^4\text{He}$ , p/d, ...)

BaBar used fused silica radiator bars,  
4.9 m (length) x 17.25 mm (thickness) x 35 mm (width).

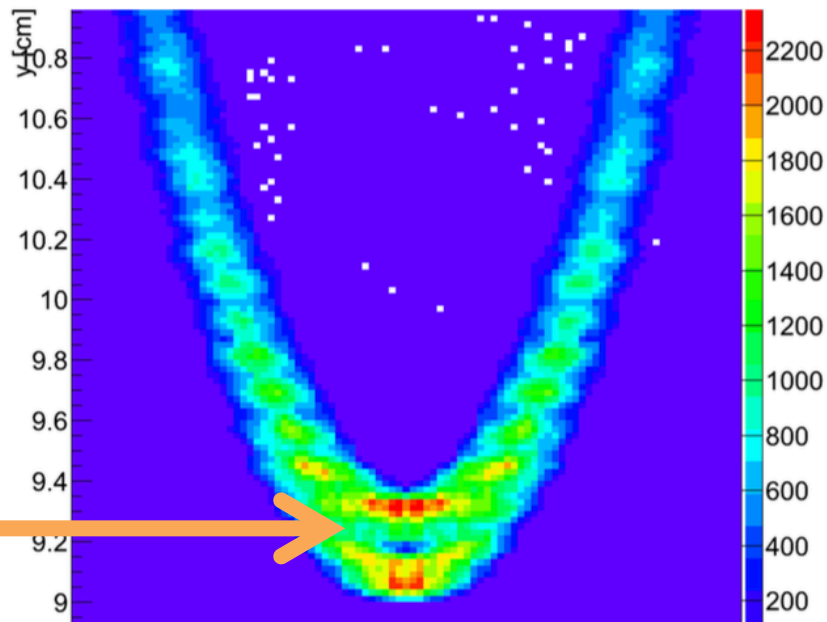
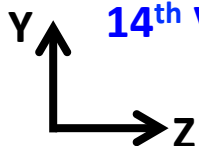


AMS-100 DIRC



- Focal Plane instrumented with SiPM arrays.

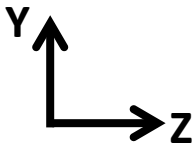
- ${}^9\text{Be}/{}^{10}\text{Be}$  at 25 GeV/c momentum from P.S. Marrocchesi, 14<sup>th</sup> VCI, Feb. 2016



A cross-sectional diagram of the Outer Detector assembly. The assembly is shown as a central white rectangular region surrounded by various colored layers. At the top and bottom, there are multiple layers of blue, grey, red, green, and orange. On the left and right sides, there are vertical stacks of blue, grey, green, and blue layers. In the center of the white region, the text 'Outer Detector' is written in blue. Below this, a table lists the components, their weights, and their X0 values. A coordinate system is shown in the bottom left corner with a vertical 'y' axis pointing up and a horizontal 'z' axis pointing right.

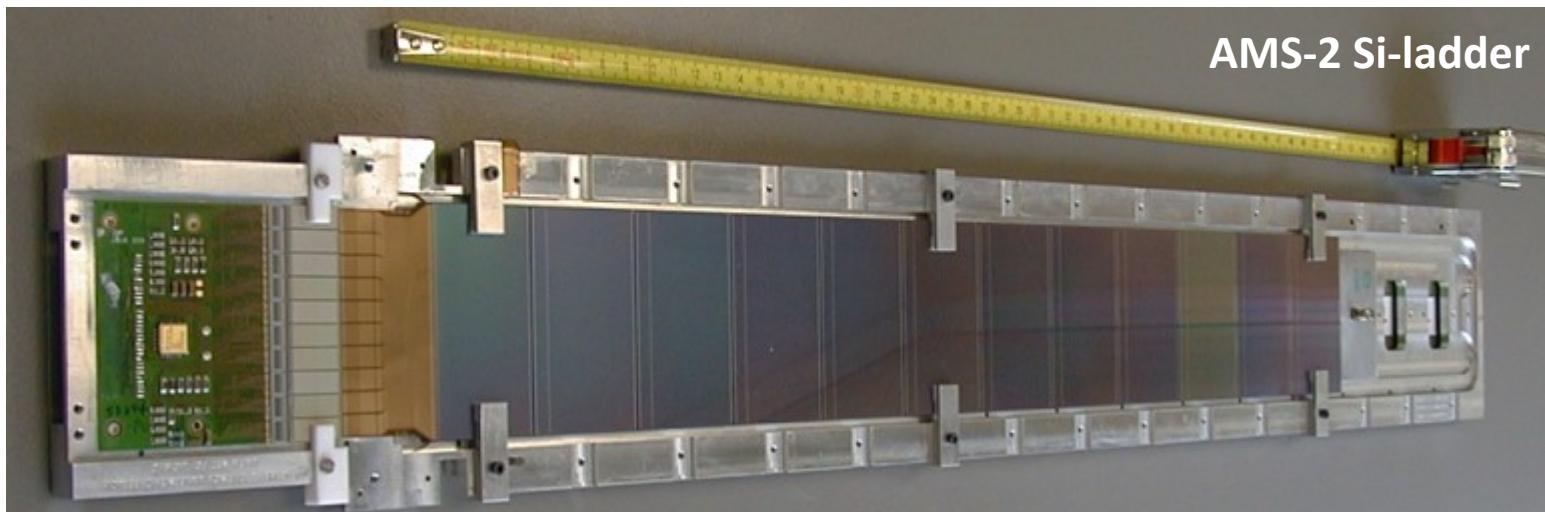
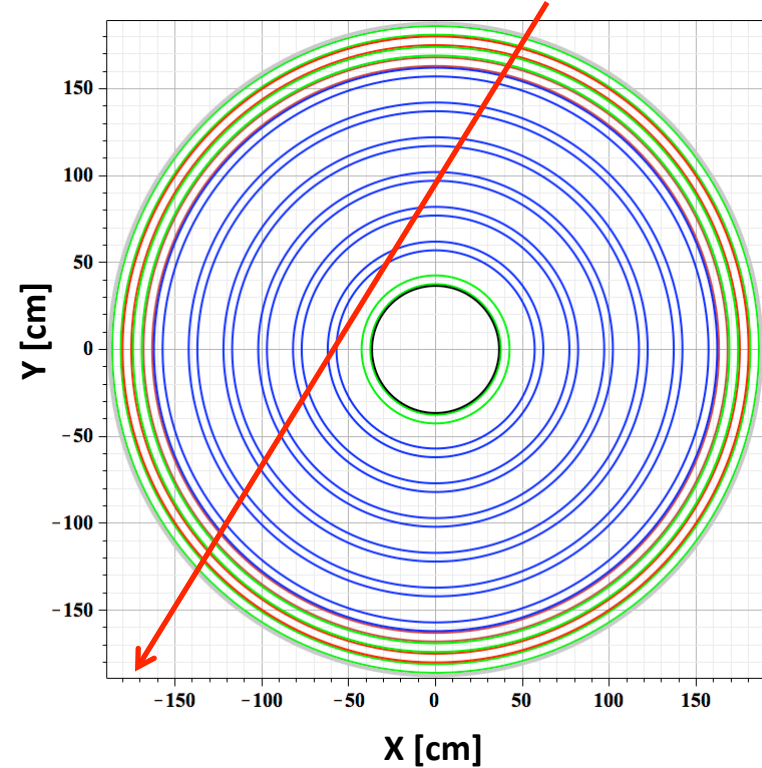
# Outer Detector

Name	Weight [kg]	X0
LEPS	1,390	0.05
ACC	1,640	0.05
Magnet	11,430	0.48
ToF	1,560	0.05
DIRC	1,950	0.14
<b>Total</b>	<b>17,970</b>	<b>0.78 (19 g/cm<sup>2</sup>)</b>



# Silicon Tracker

- Measures the particle track in the bending plane with up to 24 points.
- Readout pitch 0.1 mm, single point resolution 5  $\mu\text{m}$ .
- Single sided Si-detectors (10cm x 10cm) form a ladder of 1m length. In total 3300 ladders are needed.
- Total area 330 m<sup>2</sup>, 3.3  $10^6$  readout channels, Power consumption Front-End 1 mW/Channel  $\Leftrightarrow$  3300 W total.



# SciFi Tracker

- Measures the particle track in the non bending plane with up to 24 points.
- Readout pitch 0.25 mm, single point resolution 50  $\mu\text{m}$ .
- Constructed from 0.25 mm diameter scintillating fibers, aranged in 6 layers to form a up to 2.5m long fiber mat.
- 0.576 Million SIPM readout channels, 8000 km fibers.

Si-Ladder

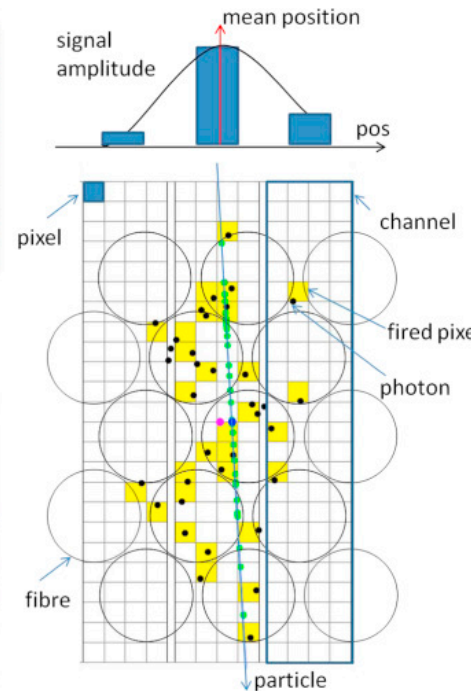
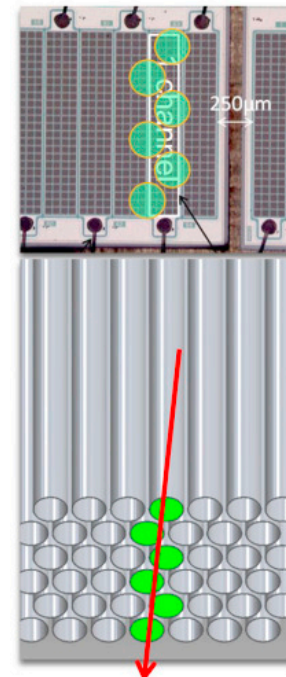
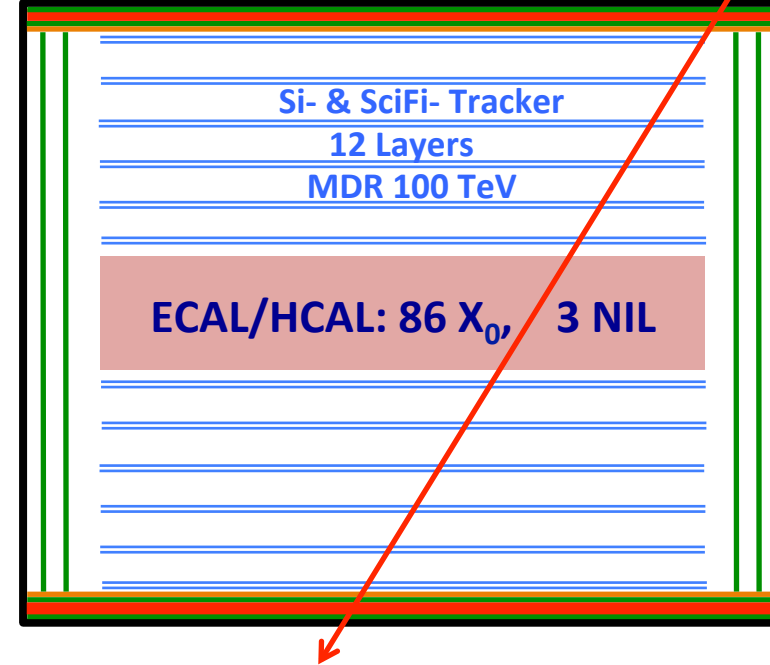
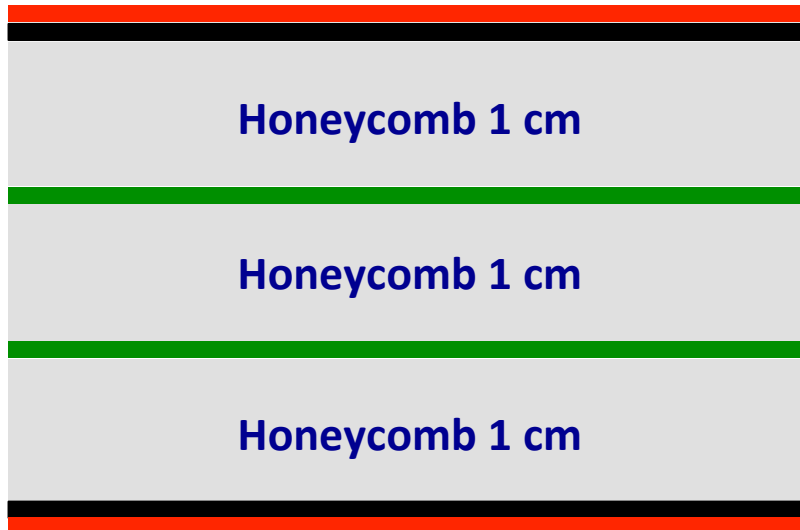
CF-Skin

SciFi-Mat

SciFi-Mat

CF-Skin

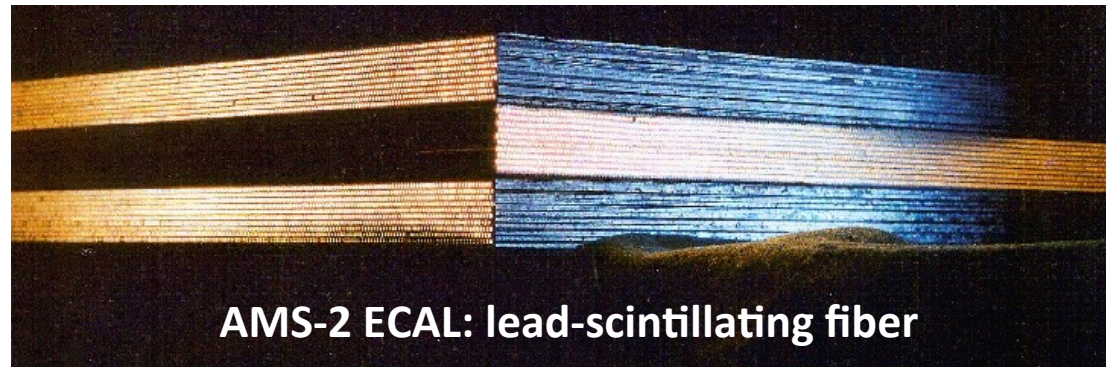
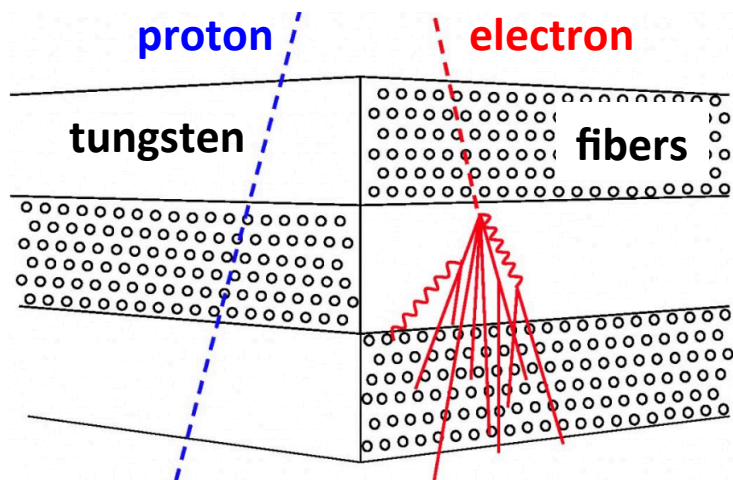
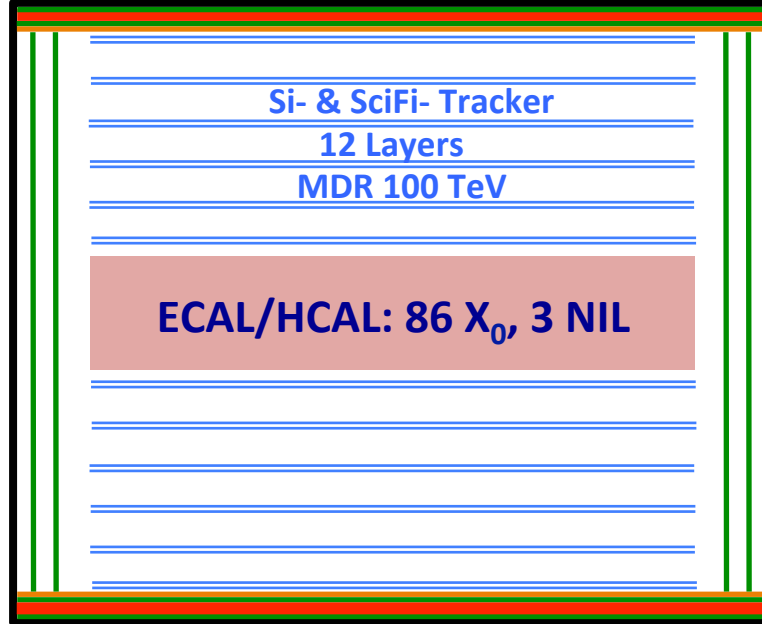
Si-Ladder



# Calorimeter System ECAL/HCAL

It is a fine grained **tungsten-scintillating fiber sampling calorimeter with SiPM readout** that allows precise, 3-dimensional imaging of the longitudinal and lateral shower development, providing high electron/hadron discrimination and good energy resolution.

- Diameter 73 cm, Length 4 m, Weight 14,550 kg, depth 86  $X_0/3$  NIL, 100,000 readout channels
- The maximum of hadron showers will be contained up to 100 TeV.



A non interacting proton passing through ECAL. The electron produces an electromagnetic shower.